

# The ALMA Antennas

From Specification to Science



Jeff Mangum

Atacama Large Millimeter/submillimeter Array  
Jansky Very Large Array  
Robert C. Byrd Green Bank Telescope  
Very Large Baseline Array



# ALMA Development Timeline

- September 10, 1982: MMA Memo #1, “The Concept of a Millimeter Array” by F.N. Owen
  - “The purpose of this report is to summarize the concept of a millimeter wave array which NRAO might build in the next 5 to 10 years.”
  - “Much of the science that a millimeter array would produce would come from spectral line observations.”
  - “The concept has started with a Wye shaped array with at least 15 antennas. Antennas of 10 meter diameter seem to be the largest practical size.”
  - “The antennas should be capable of operating at 100 GHz most of the time and at 230 GHz under the best conditions.”
  - “Wide bandwidths (1 to 5 GHz) and low system temperatures would be necessary for the desired sensitivity.”
  - “Cost Estimate for Millimeter Array:
    - Telescopes: \$11.0 M
    - Receivers: \$7.5 M
    - Correlator, IF, LO: \$4.0 M
    - Computers: \$1.5 M
    - Site: \$3 M
    - Transporters: \$1 M
    - Management: \$2 M
    - Contingency & Etc.: \$6 M
    - **TOTAL = \$36 M**



# ALMA Development Timeline (Cont.)

- February 10-11, 1983: NSF Review on the “future of millimeter astronomy” reviews NRAO contribution which describes the science that one could do with a “millimeter array”
- October 1, 1984: MMA Memo #25: “Are We Thinking Boldly Enough?” by M.A. Gordon
  - “...our millimeter array [concept] may prove to be too conventional to win funding in tight economic times.”
  - “...I argue that we should design the Array to be truly unique.”
  - “My solution is to put the array in the southern hemisphere.”
- 1985 through 1995: MMA Design Development
  - 40 8m antennas
  - Operate up to 345 GHz
- 1999: MOU between NRAO and ESO for the joint development of a millimeter array in Chile
- 2002-2004: Two prototype antennas evaluated at the VLA site
- February 25, 2003: North America / Europe bilateral agreement to construct and operate ALMA
- September 2004: North America / Europe / Japan trilateral agreement to construct and operate an enhanced Atacama Large Millimeter / submillimeter Array (ALMA)



# ALMA Key Science Goals

- Detect CO or CII in a normal galaxy like the Milky Way at a redshift of 3 in less than 24 hours
- Image the gas kinematics in protostars and protoplanetary disks around young Sun-like stars in the nearest ( $d = 150$  pc) molecular clouds
- Provide precise high-dynamic range images at an angular resolution of 0.1 arcsec



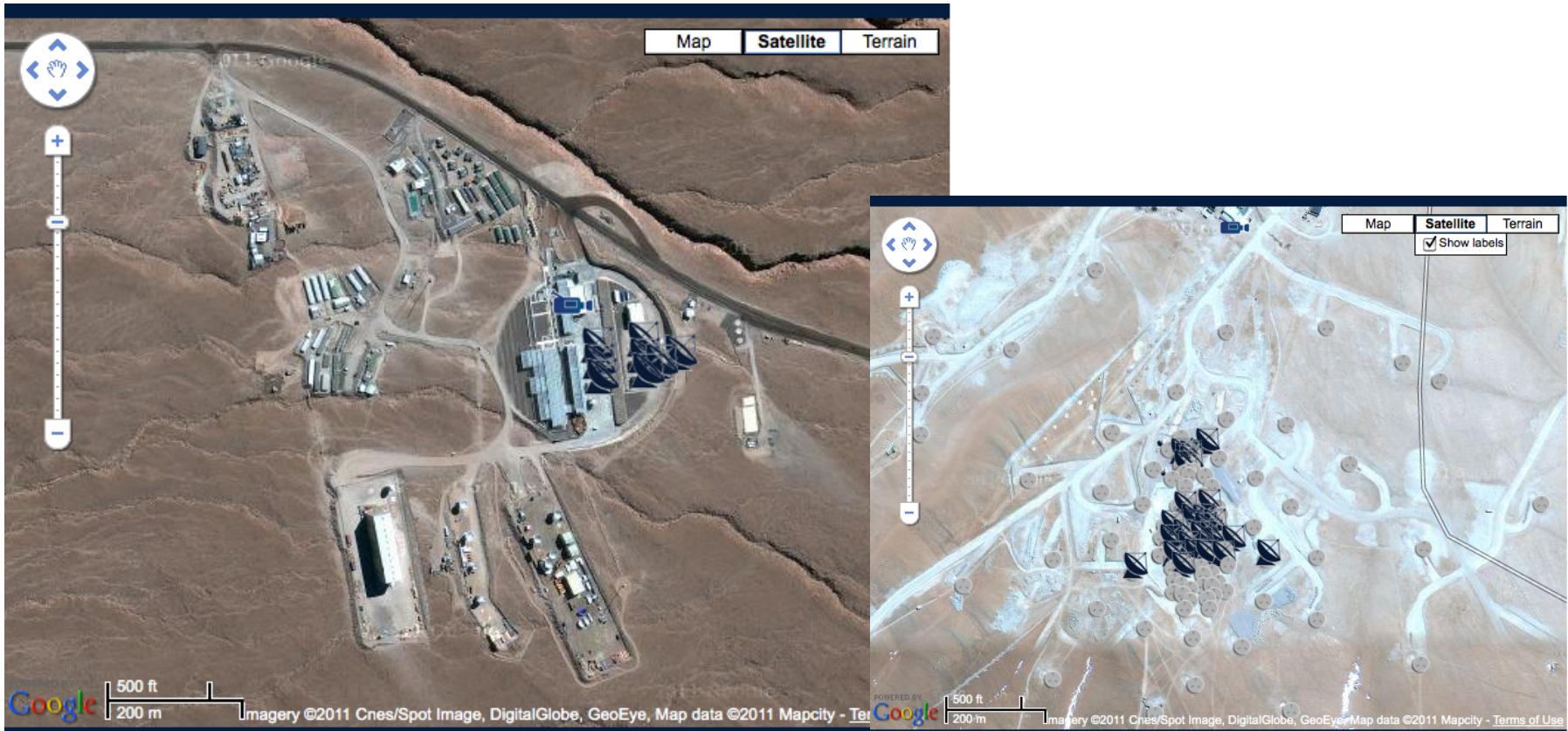
# ALMA Design

- Continuous Frequency Coverage: 100 to 1000 GHz
- >6600 m<sup>2</sup> Collecting Area
- Baselines: 15 m to 16 km
- Ability to Process 16 GHz of Bandwidth
- 24 Hour Operation
- Continuum Sensitivity: 0.05 to 1 mJy in 60 seconds
- Spectral Line Sensitivity: 7 to 62 mJy in 60 seconds at 1 km/s resolution



# ALMA Site

- Chilean Andes at altitude 5050 m (Array Operations Site: AOS)
- Operations Support Facility (OSF) at 2900 m



# A Brief History of the ALMA Antennas

- 2000: ALMA US and EU each procure prototype antennas to test viability of ALMA antenna performance specifications
  - 2000 February 18: Purchase order for VertexRSI antenna signed
- 2001 May 14: Antenna Evaluation Group (AEG) is formed
  - International group of antenna testers tasked with evaluating the prototype antennas
- 2001 June: Outfitting of ALMA Test Facility (ATF) at VLA site begins
  - 2002 November: ALMA US prototype antenna able to position in (Az,El)
- 2003 January: Testing of ALMA US prototype antenna at ATF begins
  - 2003 January: First radiometric measurements (1mm and 3mm) made with ALMA US prototype antenna
  - 2003 March: Provisional acceptance of ALMA US prototype antenna
  - 2003 April-July: Holographic measurements of ALMA US prototype antenna
  - 2003 October: Full acceptance of ALMA US prototype antenna
  - 2003 November: First astronomical image (Saturn at 3mm) with ALMA US prototype antenna
  - 2004 January: ALMA EU prototype antenna able to position in (Az,El)
- 2004 February: Testing of ALMA EU prototype antenna at ATF begins
  - 2004 January-February: Holographic measurements of ALMA EU prototype antenna
- 2004 May 28: Prototype antenna testing ends
  - 2004 May 28: Executive report describing evaluation results issued to ALMA, NRAO, and ESO management
  - 2004 December 12: Final reports describing surface accuracy, pointing, and fast switching performance of ALMA US and EU prototype antennas submitted
- 2004: Production antenna contracts chosen by US and EU



# ALMA Antennas

- 54 12m antennas
- ACA adds 12 7m antennas
- Performance Requirements
  - Surface Accuracy:  $25\mu\text{m}$  RMS ( $20\mu\text{m}$  goal)
  - Absolute Pointing Accuracy: 2 arcsec all-sky
  - Offset Pointing Accuracy: 0.6 arsec over 2 degree radius
  - Fast Switching
    - 1.5 degree move in 1.5 seconds
    - Settle to 3 arcsec peak pointing error at 1.5 seconds after start of switch
    - Settle to 0.6 arcsec RMS tracking error over 2 to 4 seconds after start of switch
- Path Length Stability:  $15\mu\text{m} / 20\mu\text{m}$  (non-repeatable / repeatable)
- Primary Operating Conditions
  - $T_{\text{amb}} = -20 \text{ to } +20 \text{ C}$
  - $\Delta T_{\text{amb}} \leq 0.6 / 1.8 \text{ C}$  over 10 / 30 minute durations
  - $V_{\text{wind}} \leq 6 / 9 \text{ m/s}$  (day / night)



# ALMA Key Science Goals Rely on High Performance Antennas

Key Science Goal	Antenna Spec
High Dynamic Range Imaging at 0.1 arcsec resolution	Pointing, Surface, Path Length, Fast Switching
Detect High-Redshift Galaxies	Pointing, Surface, Path Length, Fast Switching
Imaging at 100 to 1000 GHz	Pointing, Surface, Path Length, Fast Switching



# ALMA Antennas Delivery Status

- 41 antennas delivered and integrated
  - North American (Vertex): 21 of 25 (last antenna delivered by 2012-10-01)
  - European (AEM): 7 of 25
  - East Asian (Melco): 4 of 4 (12m) and 9 of 12 (7m)
- 36 antennas in use at the AOS (April 2012)



# Vertex and AEM Production Antennas

<b>Property</b>	<b>Vertex</b>	<b>AEM</b>
Base/Yoke	Insulated Steel	Insulated Steel
BUS	Al honeycomb with CFRP plating, 24 sectors, open back, covered with removable GFRP sunshades	Solid CFRP plates, 16 closed-back sectors glued and bolted together
Receiver Cabin	Insulated steel with INVAR cone interface to BUS	CFRP; direct-connection cabin to BUS
Base	3-point support; bolt connection with foundation	3-point support; bolt connection with foundation
Drive	Gear and pinion	Direct-drive with linear motors
Brakes	Integrated on servo motor	Hydraulic disk
Encoders	Absolute (BEI)	Incremental (Heidenhain)
Panels	264 panels, 8 rings, machined Al, open-back, 8 adjusters (3 lateral/5 axial) per panel	120 panels, 5 rings, Al honeycomb with replicated Ni skins. Rh coated, 5 adjusters per panel
Apex/Quadripod	CFRP structure, "+" configuration	CFRP structure, "x" configuration
Focus Mechanism	Hexapod (5 DOF)	Hexapod (5 DOF)
Total Mass	~107 tonnes	~100 tonnes
Mass Dist. (El/Az)	50%/50%	35%/65%



# Early Science

The Antennae Galaxies



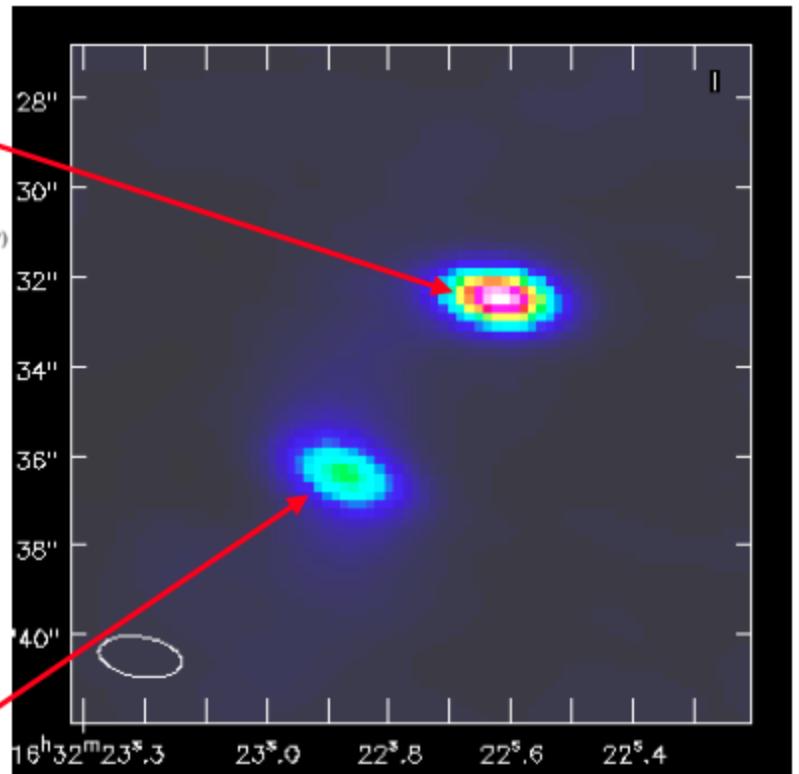
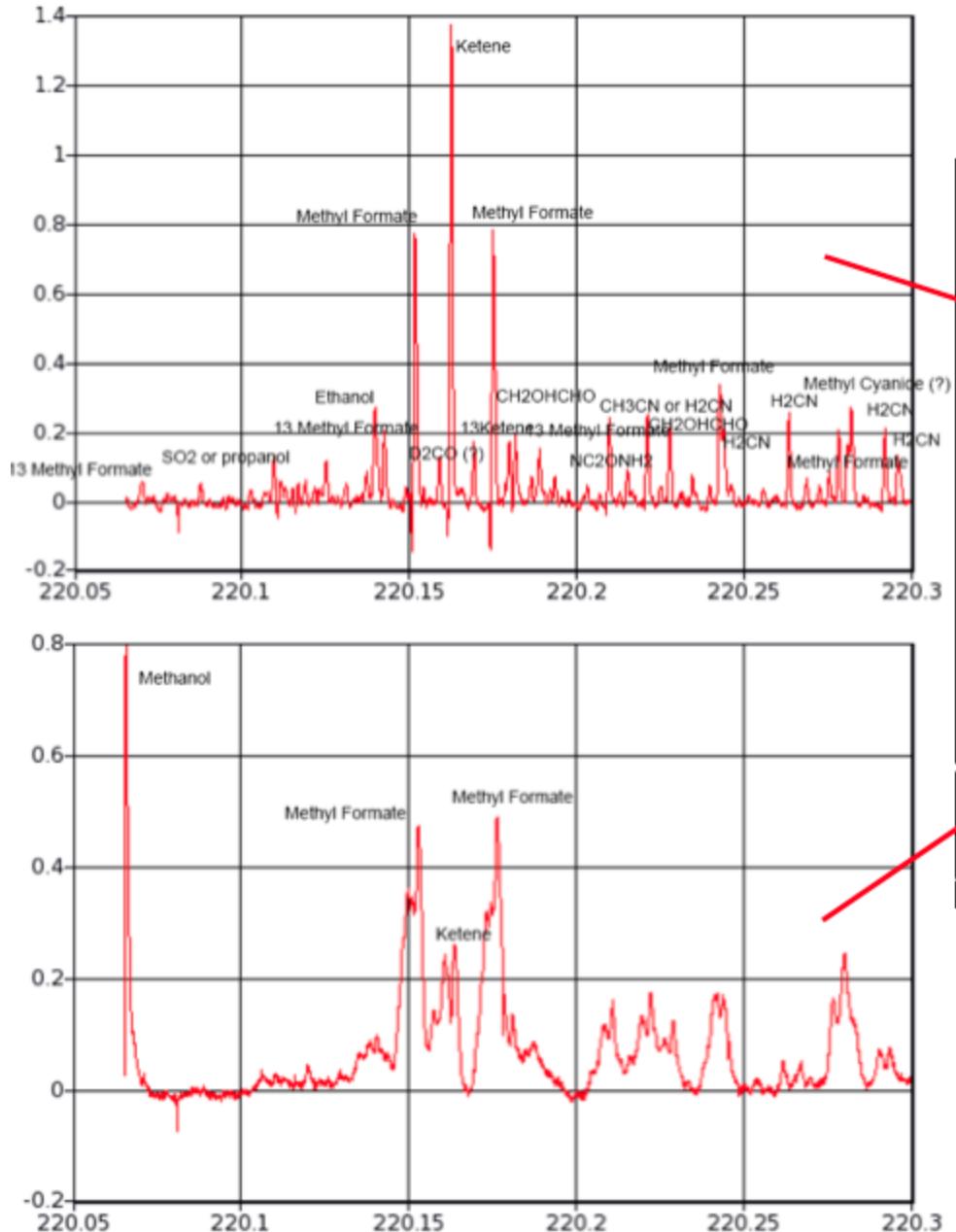
Blue = visible (HST)

Red = CO 1-0 (Band 3)

Yellow = CO 3-2 (Band 7)

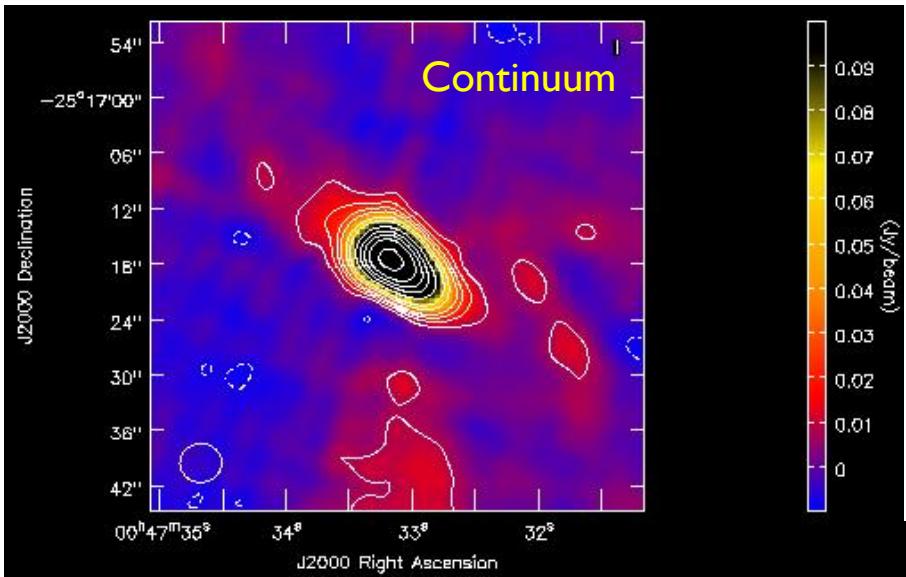


# Early Science

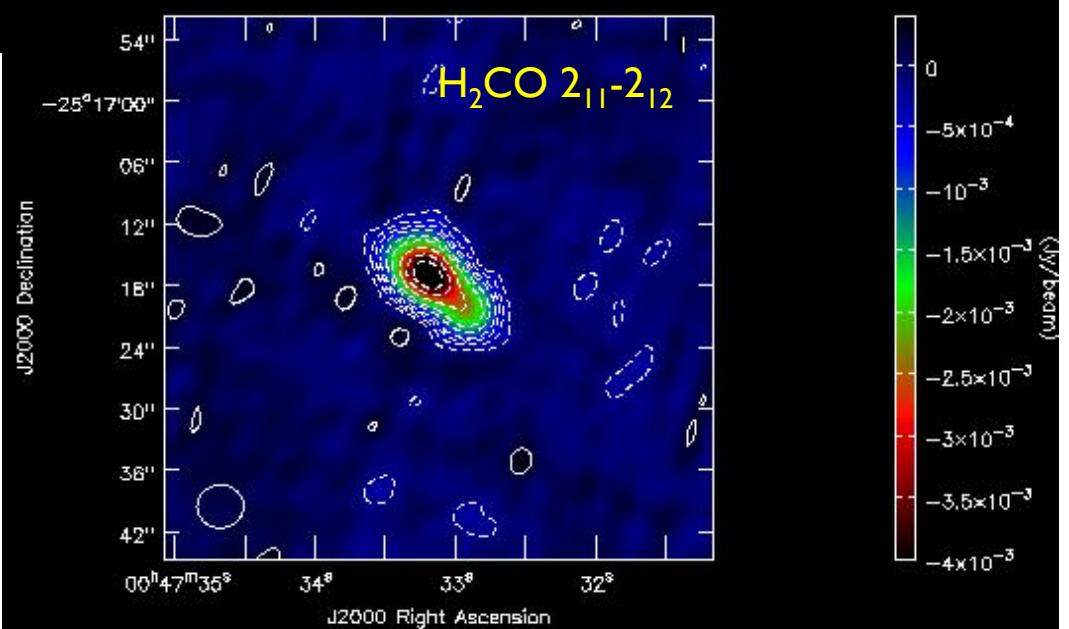
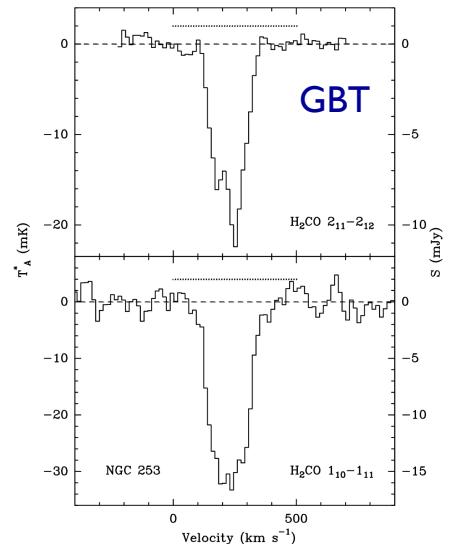


IRAS16293

# Why Pointing Is Important



NGC253  
 $\text{H}_2\text{CO } 2_{11}-2_{12}$  (14.48 GHz)  
 JVLA DnC Array ( $\theta_b = 5$  arcsec)  
 FWHM  $\cong 200$  arcsec



## Pointing Errors:

- Limit dynamic range to i.e. < 300:I when  $\sigma_p \cong 0.02$  FWHM (0.6 arcsec at 230 GHz; Braun 1989, MMA Memo 54)
- “Blurr” images
- Re-distribute flux from compact to extended structures
- JVLA at 2 cm:  $0.02 * \text{FWHM} \cong 4$  arcsec

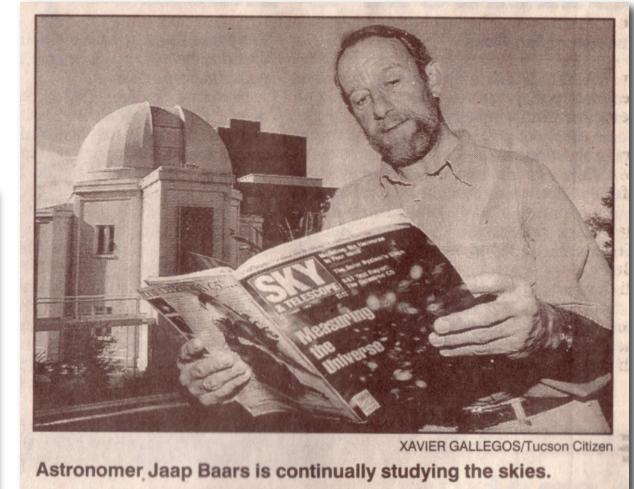


# The ALMA Test Facility (ATF)



# The AEG

- Jeff Mangum (NRAO)
- Jaap Baars (ESO/MPIfR)
- Albert Greve (IRAM)
- Robert Lucas (IRAM)
- Ralph Snel (Lund University)
- Pat Wallace (RAL)



# Jack Meadows



# ALMA NA Antenna IPT

## Antenna Testing Division

- Art Symmes (NA IPT Lead)
- Nicholas Emerson
- Kevin Flaim
- David Hunter
- Martin Mundnich
- Derek Harris
- Tony Rodriguez



# Pointing

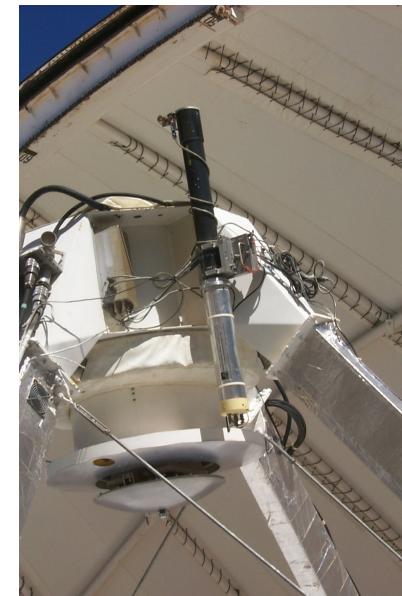
Techniques and Systems Used:

- Optical Pointing Telescope (OPT)
- 1 and 3 mm Radiometry
- Accelerometer System



# Optical Pointing Telescopes

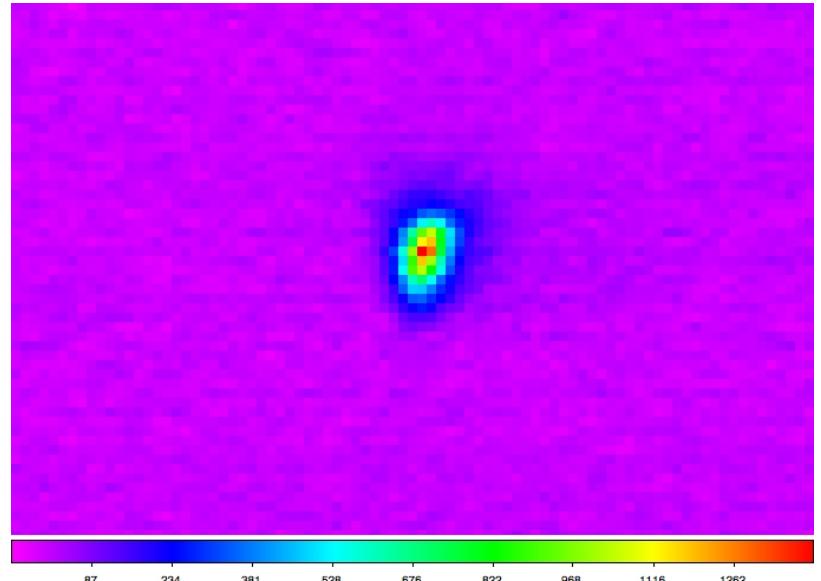
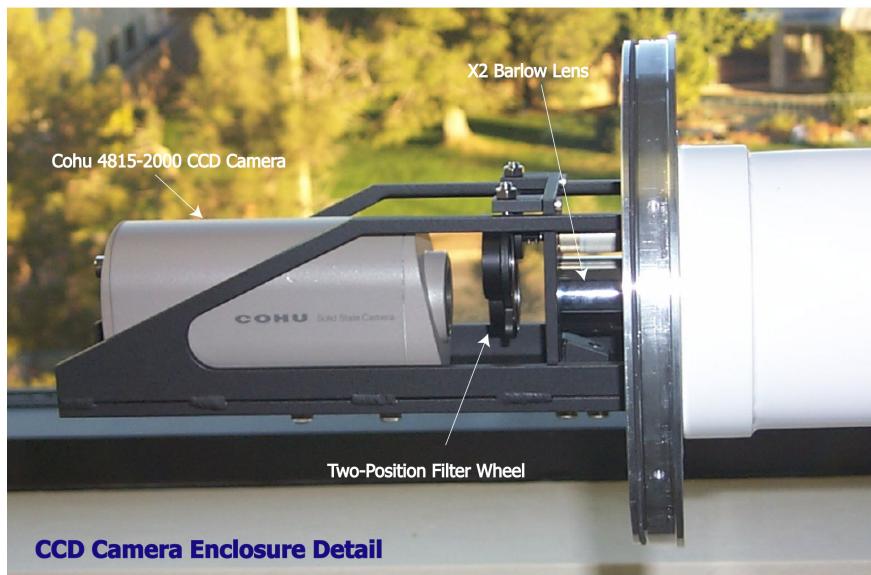
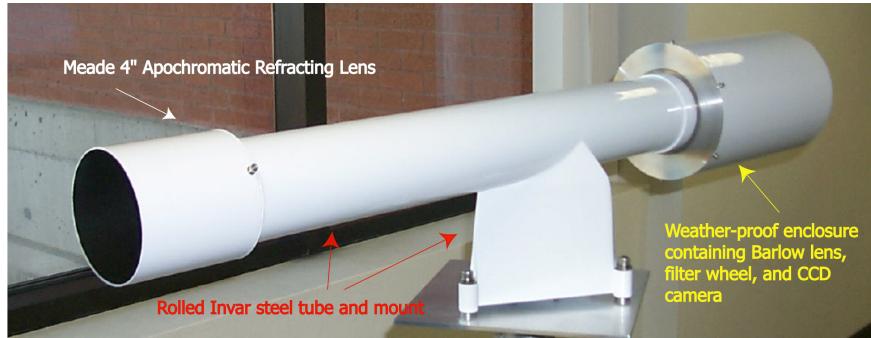
- Prototype OPTs developed at NRAO Tucson 2000-2002
  - Based on NRAO 12m and Hat Creek OPT systems
  - Two OPTs constructed and used to characterize prototype antennas 2002-2004
  - Used to characterize pointing performance on production antennas:
    - DV01-DV09
    - DA41-DA49
- Production OPTs
  - Based on prototype OPT experience
  - ALMA specification
  - Contracted to ACE (Tucson)
  - Two systems delivered to-date



NRAO 12m OPT



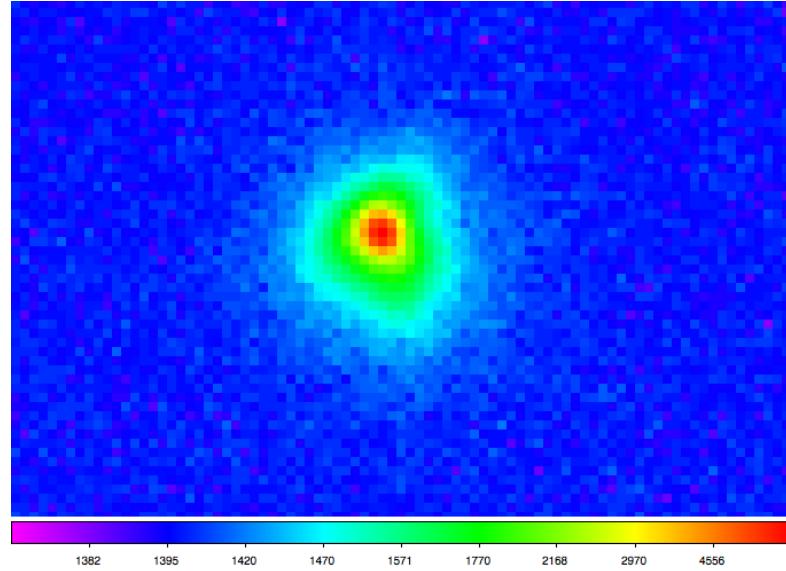
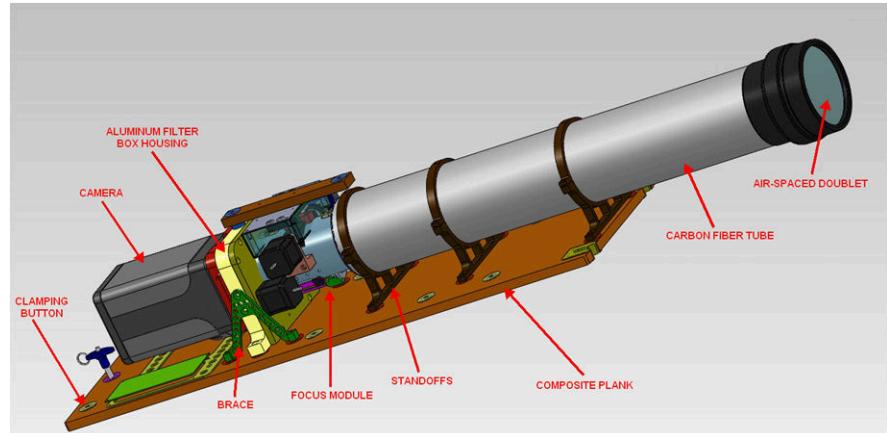
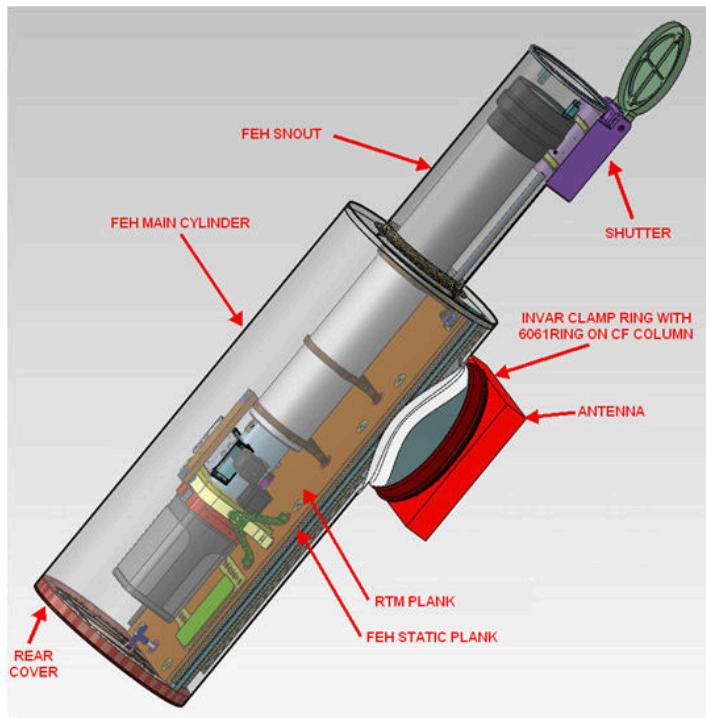
# Prototype Optical Pointing Telescopes



- $D = 4$  inches (Meade 102ED)
- $f = 920$  mm
- Effective plate scale = 1.12 arcsec
- FOV = 12 x 9 arcmin
- Limiting magnitude = 9 (in 5 sec)

# Production Optical Pointing Telescopes

- Orion 120mm f7.5 lens ( $f = 900\text{mm}$ )
- 5x Barlow lens
- Princeton Instruments Photomax 1024 CCD camera
- FOV  $\approx 8.5 \times 8.5 \text{ arcmin}$
- Effective plate scale  $\approx 0.5 \text{ arcsec}$
- Limiting magnitude  $> 10$  (in 1 second)

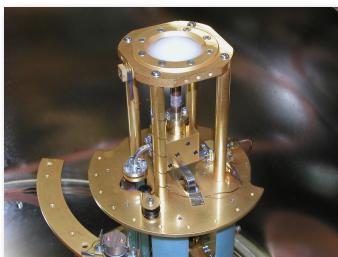


# Production Optical Pointing Telescopes



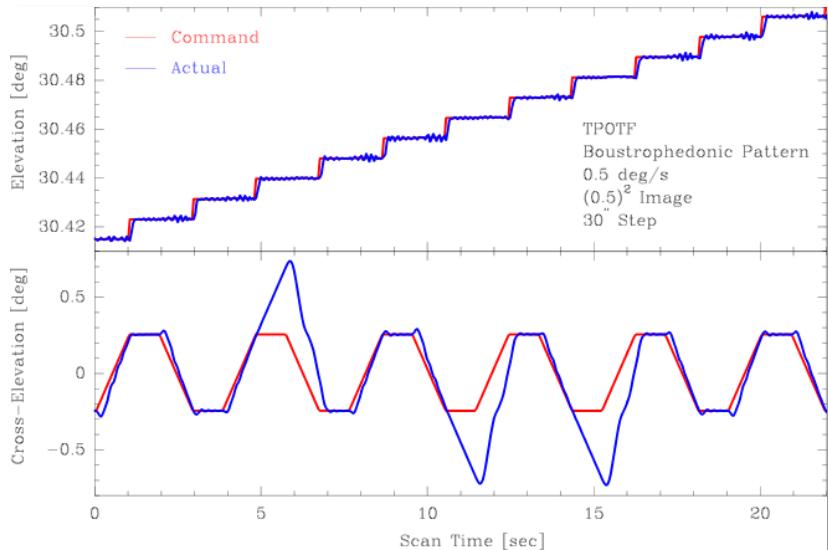
Production OPT installed in the BUS of the ALMA NA Production Antenna

# Evaluation Frontend System



# CANAnalyzer

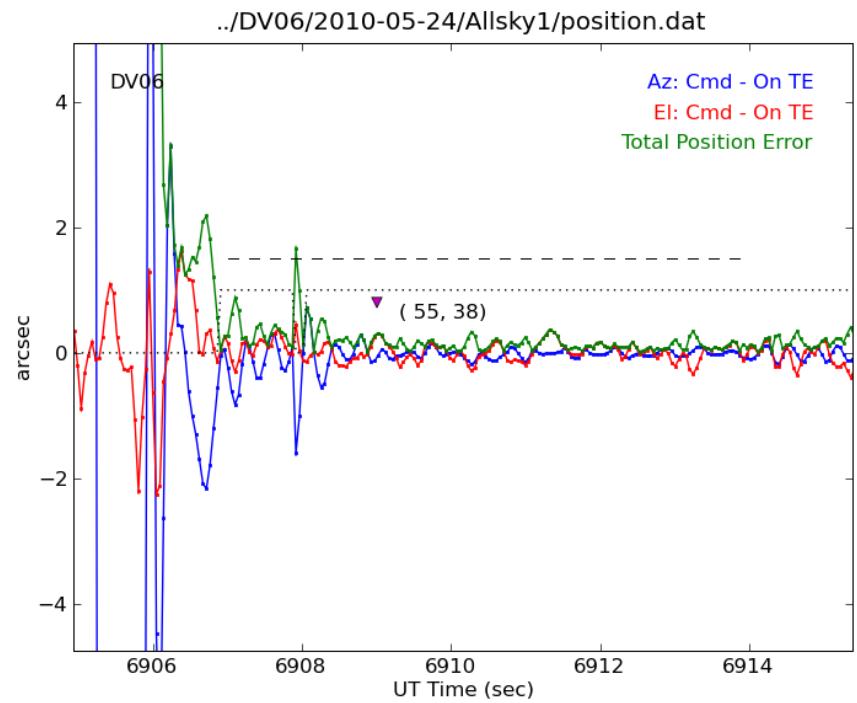
- Developed by Fritz Stauffer at ATF
- Monitors command/response antenna position each TE (48 ms)



ATF

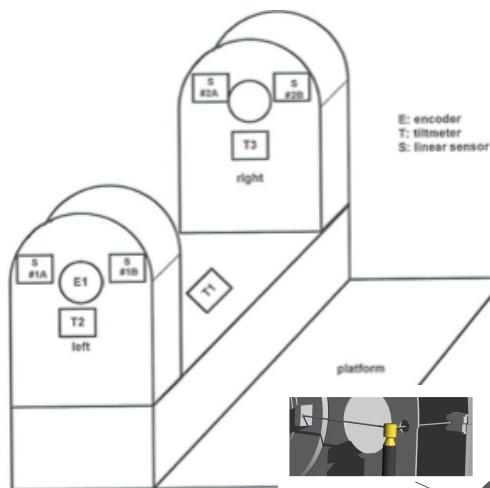


ALMA



# Vertex Metrology System

## Tiltmeter(s) and Linear Sensors



22452 EI-Encoder Compensating Device

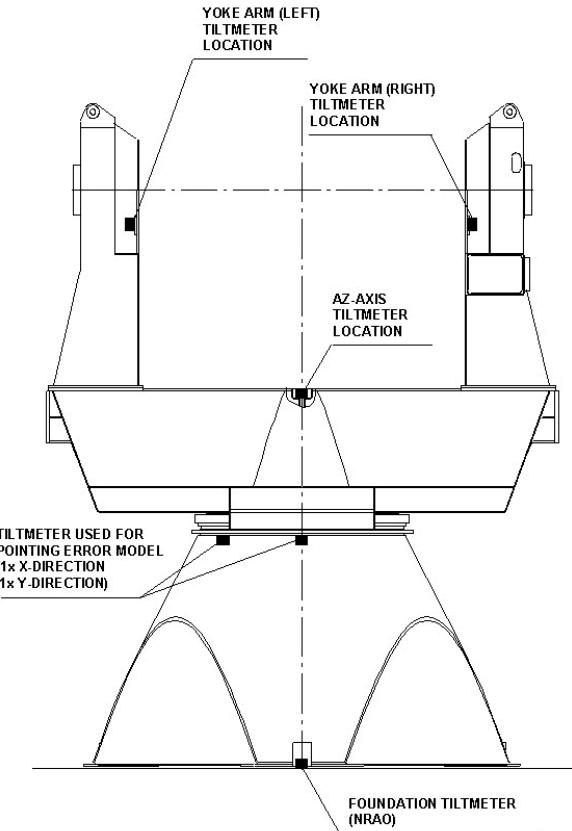
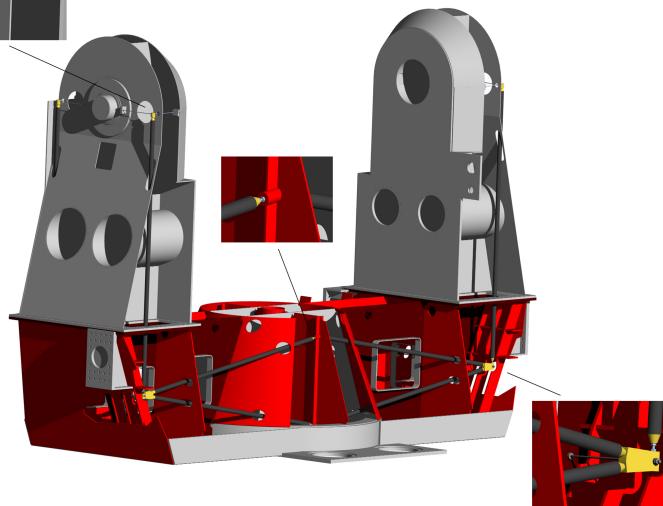
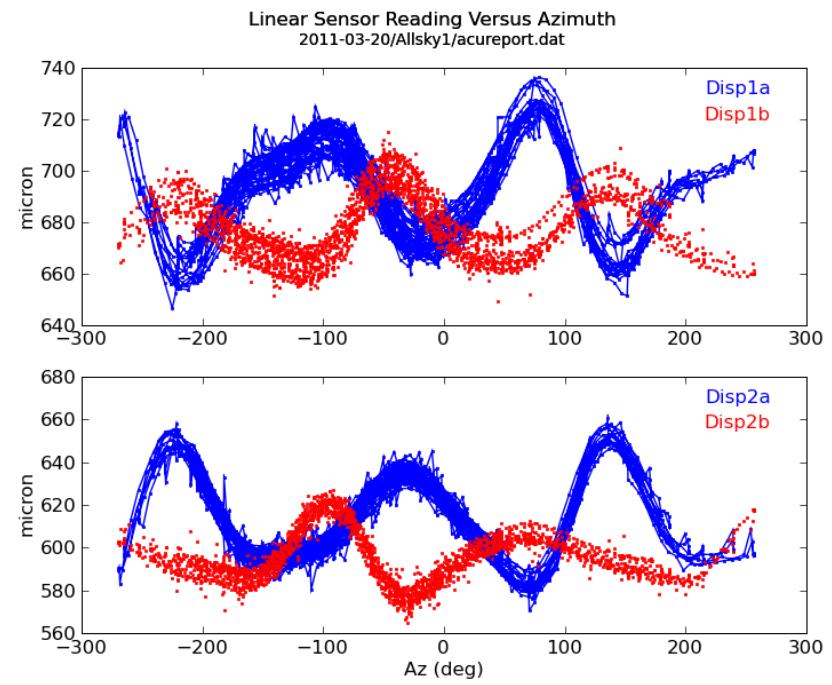
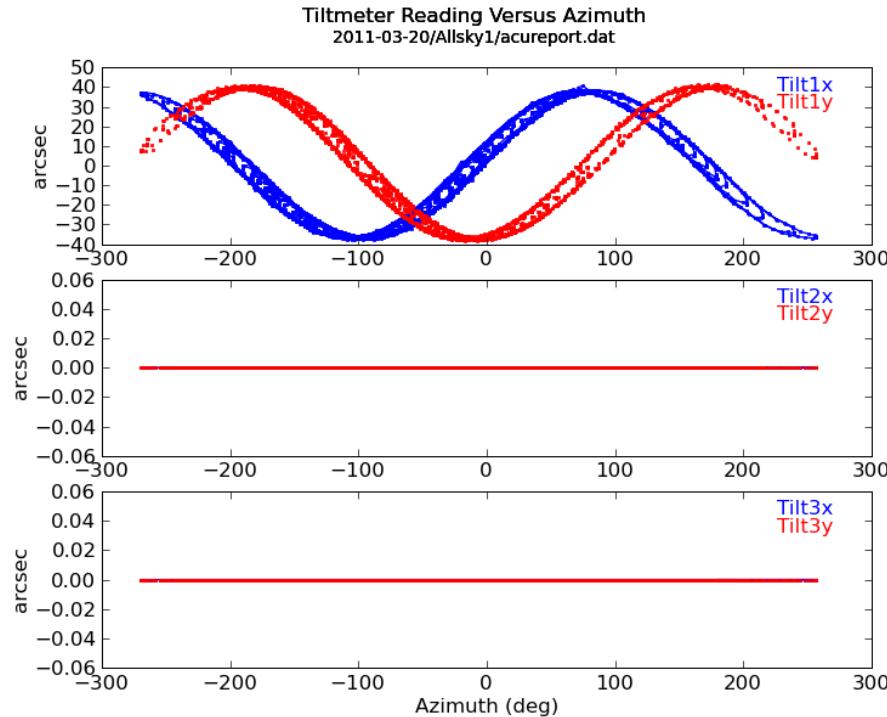


Figure: Tiltmeter Mounting Locations

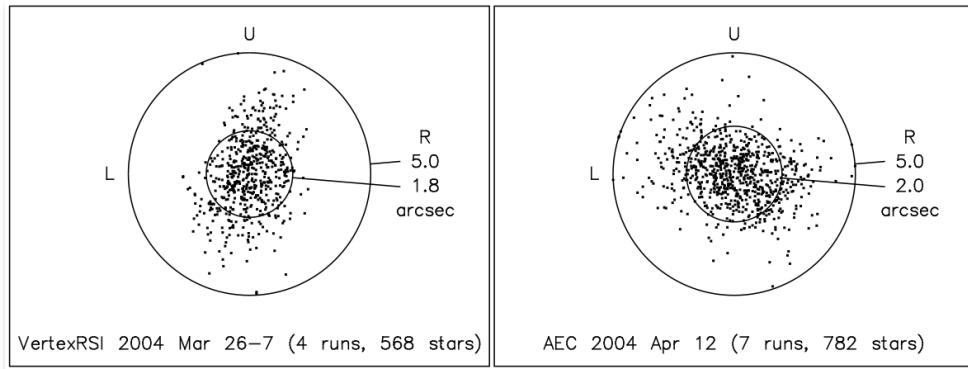
# Metrology System



# All-Sky Optical Pointing Results

## Prototype Antennas

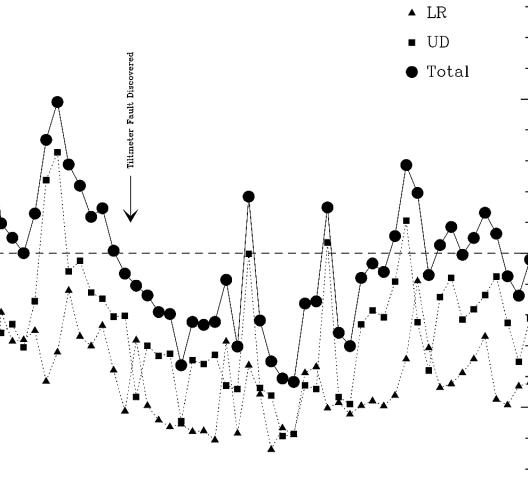
**VertexRSI**



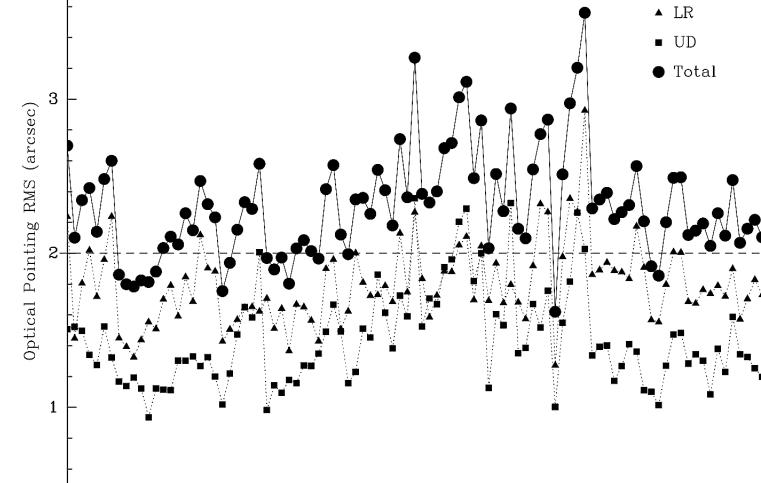
**AEC**

Optical Pointing RMS (arcsec)

VertexRSI Optical Pointing Summary



AEC Optical Pointing Summary



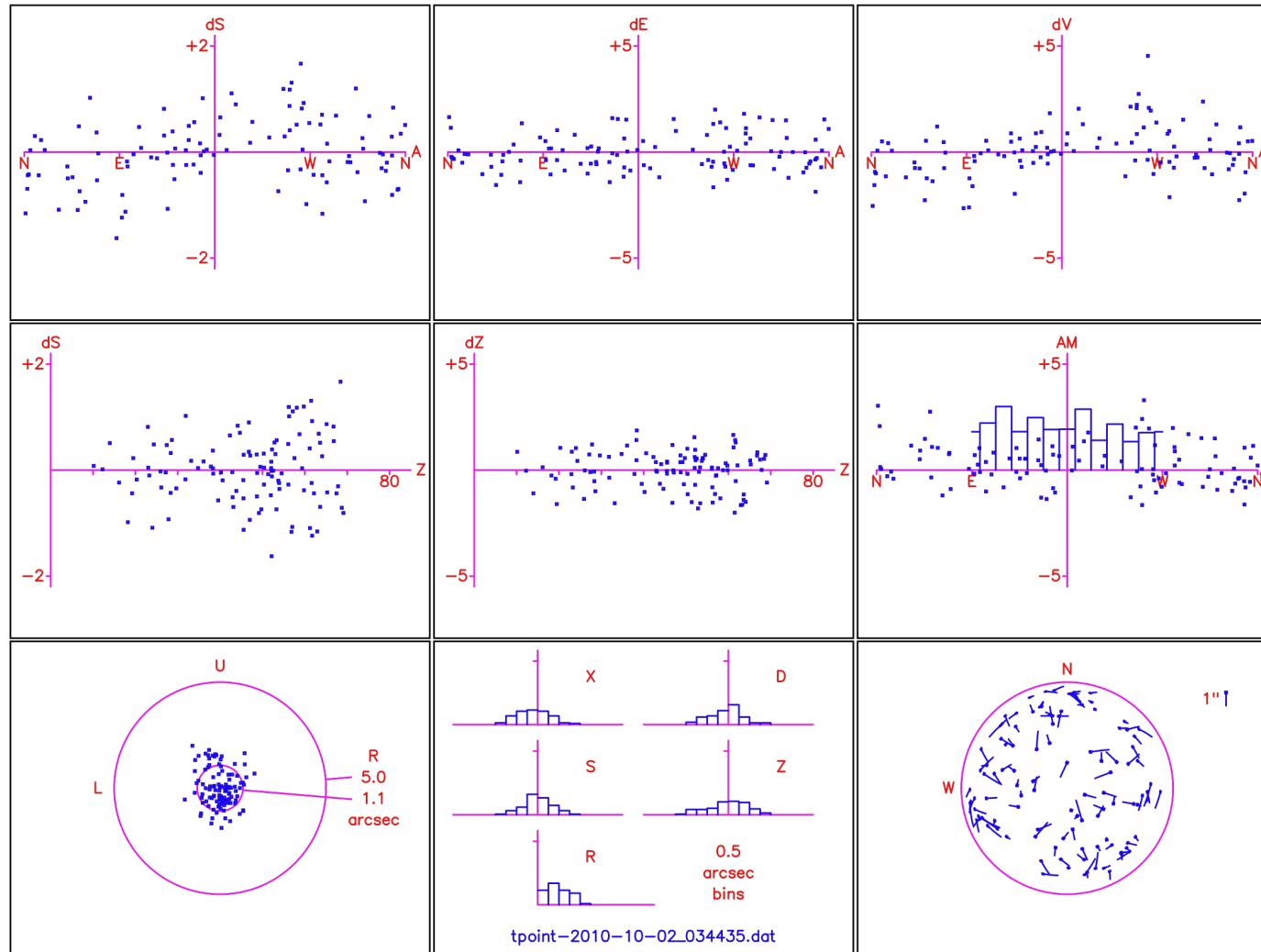
# Pointing Performance Summary

## Prototype Antennas

<b>System/Technique</b>	<b>AEC (arcsec)</b>		<b>VertexRSI (arcsec)</b>	
	<b>Absolute</b>	<b>Offset</b>	<b>Absolute</b>	<b>Offset</b>
Optical	2.0-2.6	0.3-0.8	1.3-1.8	0.3-1.1
Radiometric	<4	<0.5	<4	<0.9
Accelerometer (10 sec)	...	0.29±0.09 (spread) ±0.05 (wind)	...	0.58±0.15 (spread) ±0.08 (wind)
Accelerometer (15 min)	...	0.50±0.13 (spread) ±0.02 (extrapolation) ±0.11 (wind)	...	0.94±0.26 (spread) ±0.05 (extrapolation) ±0.20 (wind)

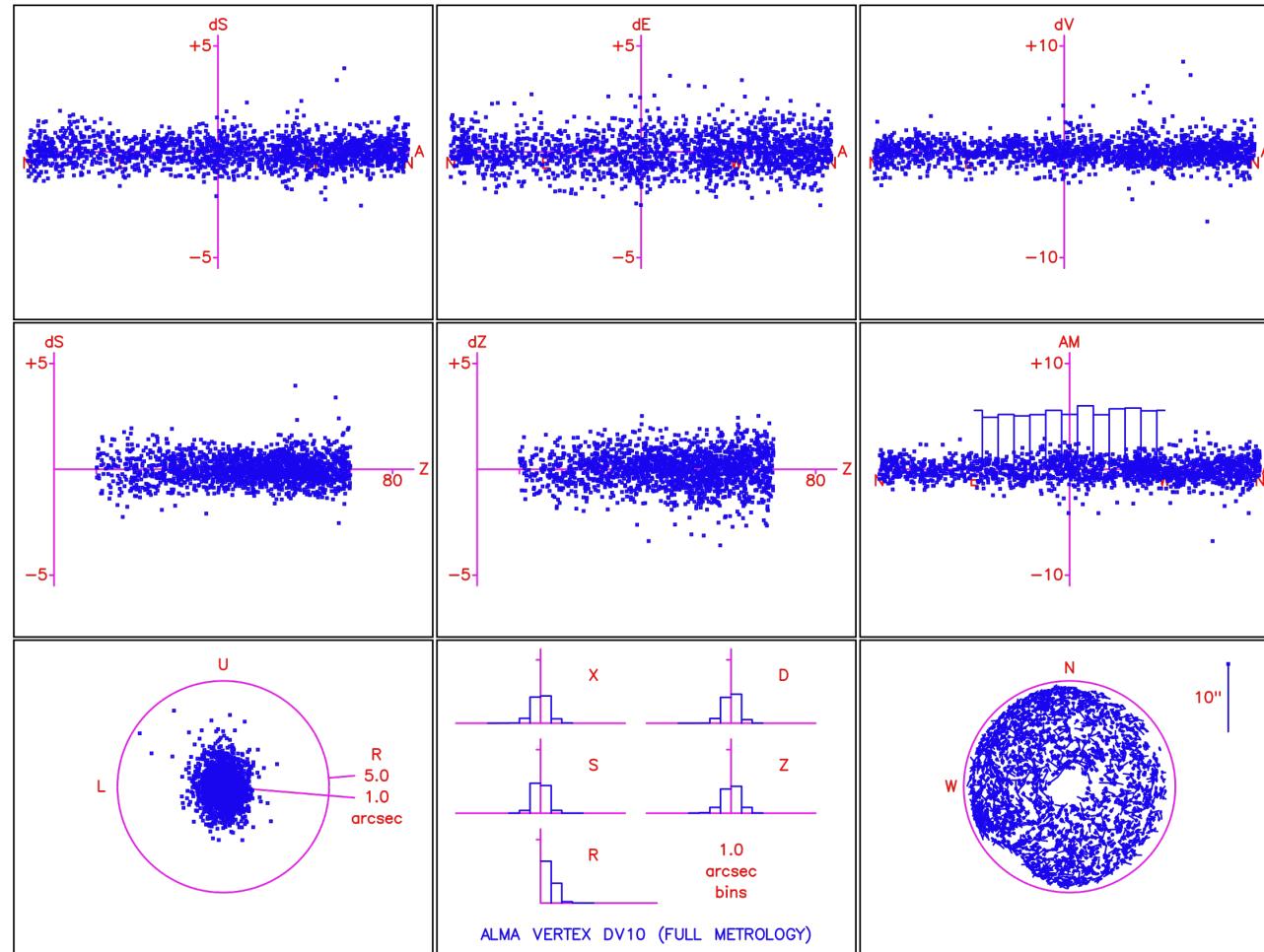


# Sample All-Sky Pointing Measurement



Production Antennas

# Cumulative All-Sky Pointing Measurement Residual for DV10

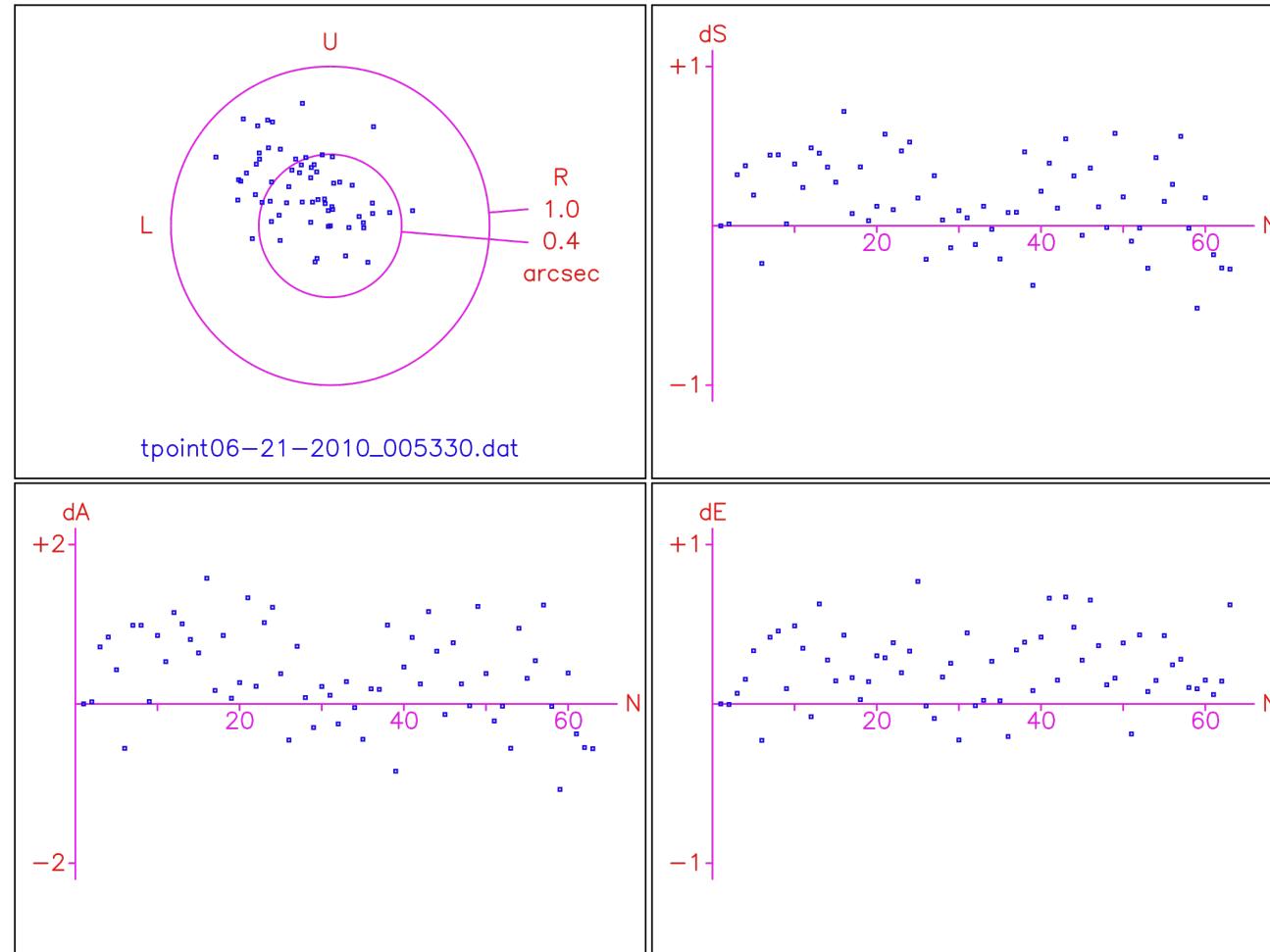


Production Antennas

NTC March 14, 2012

59

# Sample Offset Pointing Measurement Residual



Production Antennas

NTC March 14, 2012

61

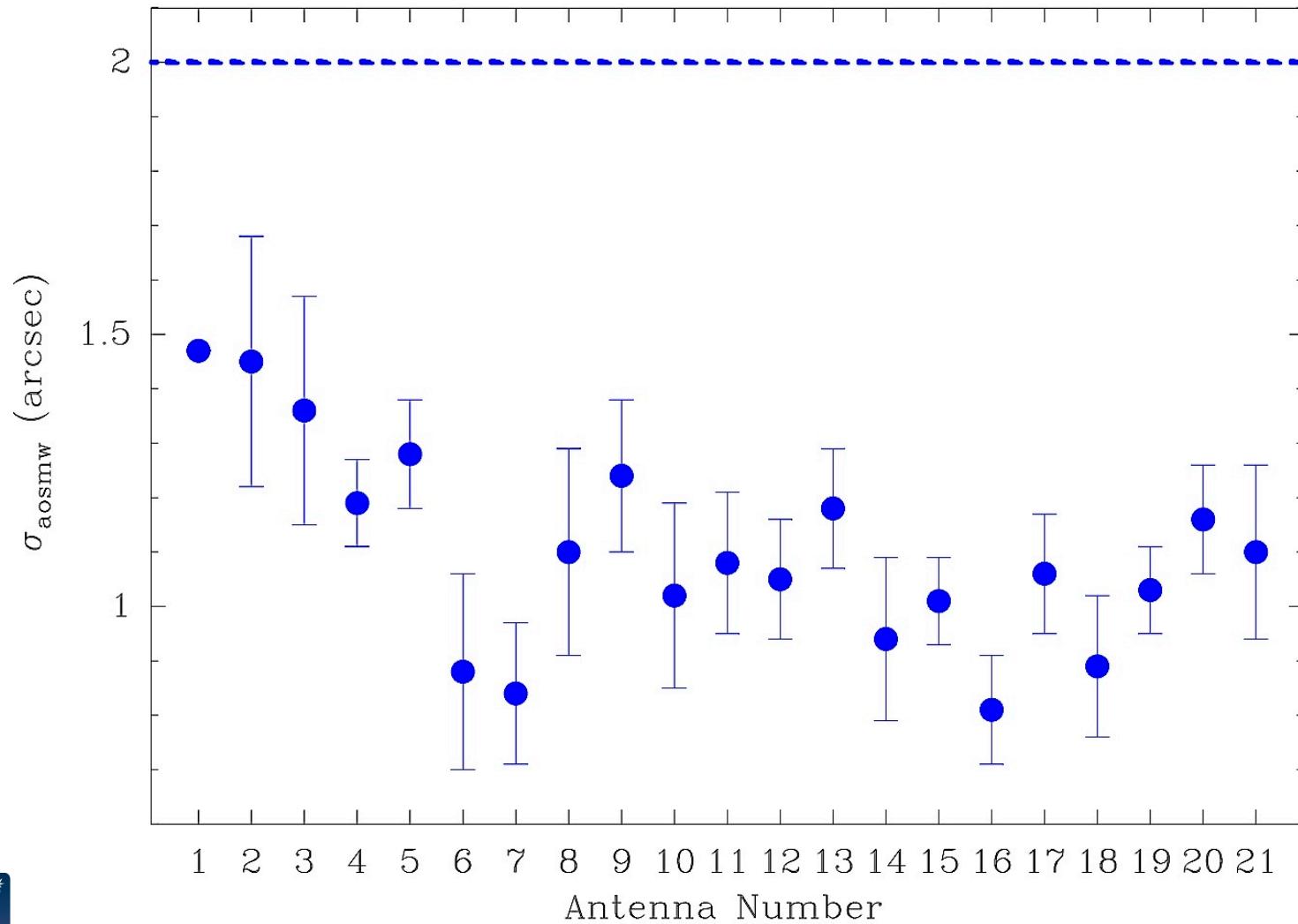
## Vertex Production Antenna All-Sky Pointing Performance

Antenna	Date <sup>a</sup>	N <sub>meas</sub>	T <sub>amb</sub> (C)	ΔT (C)	W <sub>s</sub> (m/s)	σ <sub>meas</sub> (arcsec)	σ <sub>aosmw</sub> (arcsec)
DV01	2009/01/15	57	+8.0 – +22.6	0.3 – 4.2	0.1 – 5.0	1.13	1.47
DV02	2009/04/24	21	+6.4 – +17.0	0.3 – 2.4	0.7 – 3.6	1.35±0.24	1.45±0.23
DV03	2009/09/01	22	+3.5 – +12.2	0.3 – 4.2	1.8 – 5.9	1.38±0.20	1.36±0.21
DV04	2010/02/01	18	+13.8 – +17.3	0.4 – 1.7	1.3 – 4.9	1.12±0.08	1.19±0.08
DV05	2009/10/26	18	+5.0 – +14.5	0.4 – 2.6	1.2 – 4.4	1.22±0.10	1.28±0.10
DV06	2010/06/09	12	+8.7 – +13.3	0.1 – 2.1	0.8 – 8.1	0.94±0.10	0.88±0.18
DV07	2010/04/19	26	+9.4 – +18.4	0.2 – 1.7	1.2 – 5.1	0.77±0.12	0.84±0.13
DV08	2010/07/03	18	+6.7 – +14.2	0.4 – 2.5	1.4 – 5.0	1.09±0.17	1.10±0.19
DV09	2010/07/21	15	-0.3 – +11.1	0.3 – 2.2	1.3 – 2.2	1.21±0.14	1.24±0.14
DV10	2010/10/11	19	+5.8 – +12.5	0.0 – 1.9	2.0 – 6.4	1.01±0.16	1.02±0.17
DV11	2010/12/09	19	+7.4 – +15.1	0.2 – 1.4	1.5 – 4.7	1.06±0.13	1.08±0.13
DV12	2011/01/30	14	+11.3 – +16.0	0.5 – 1.5	1.3 – 6.1	1.00±0.09	1.05±0.11
DV13	2011/03/23	22	+10.8 – +15.2	0.3 – 1.7	1.7 – 5.9	1.14±0.10	1.18±0.11
DV14	2011/05/31	25	+8.9 – +18.0	0.2 – 2.1	2.6 – 6.7	0.91±0.13	0.94±0.15
DV15	2011/07/29	8	+6.0 – +11.7	0.5 – 2.2	1.6 – 5.8	1.11±0.07	1.01±0.08
DV16	2011/08/26	19	+7.8 – +14.5	0.5 – 2.1	3.0 – 5.9	0.82±0.08	0.81±0.10
DV17	2011/10/10	19	+11.2 – +17.9	0.3 – 2.7	1.9 – 6.5	1.02±0.10	1.06±0.11
DV18	2011/12/06	18	+10.8 – +17.4	0.4 – 2.2	1.0 – 5.4	0.83±0.11	0.89±0.13
DV19	2012/01/11	14	+10.0 – +17.7	0.2 – 1.1	0.8 – 4.6	0.96±0.08	1.03±0.08
DV20	2012/04/09	18	+12.4 – +16.7	0.4 – 2.4	1.5 – 5.3	1.10±0.11	1.16±0.10
DV21	2012/05/04	23	+8.1 – +15.1	0.3 – 1.5	1.2 – 5.8	1.07±0.15	1.10±0.16

<sup>a</sup> Date of final report

# Vertex Production Antenna All-Sky Pointing Performance

Vertex Antenna All-Sky Pointing Performance

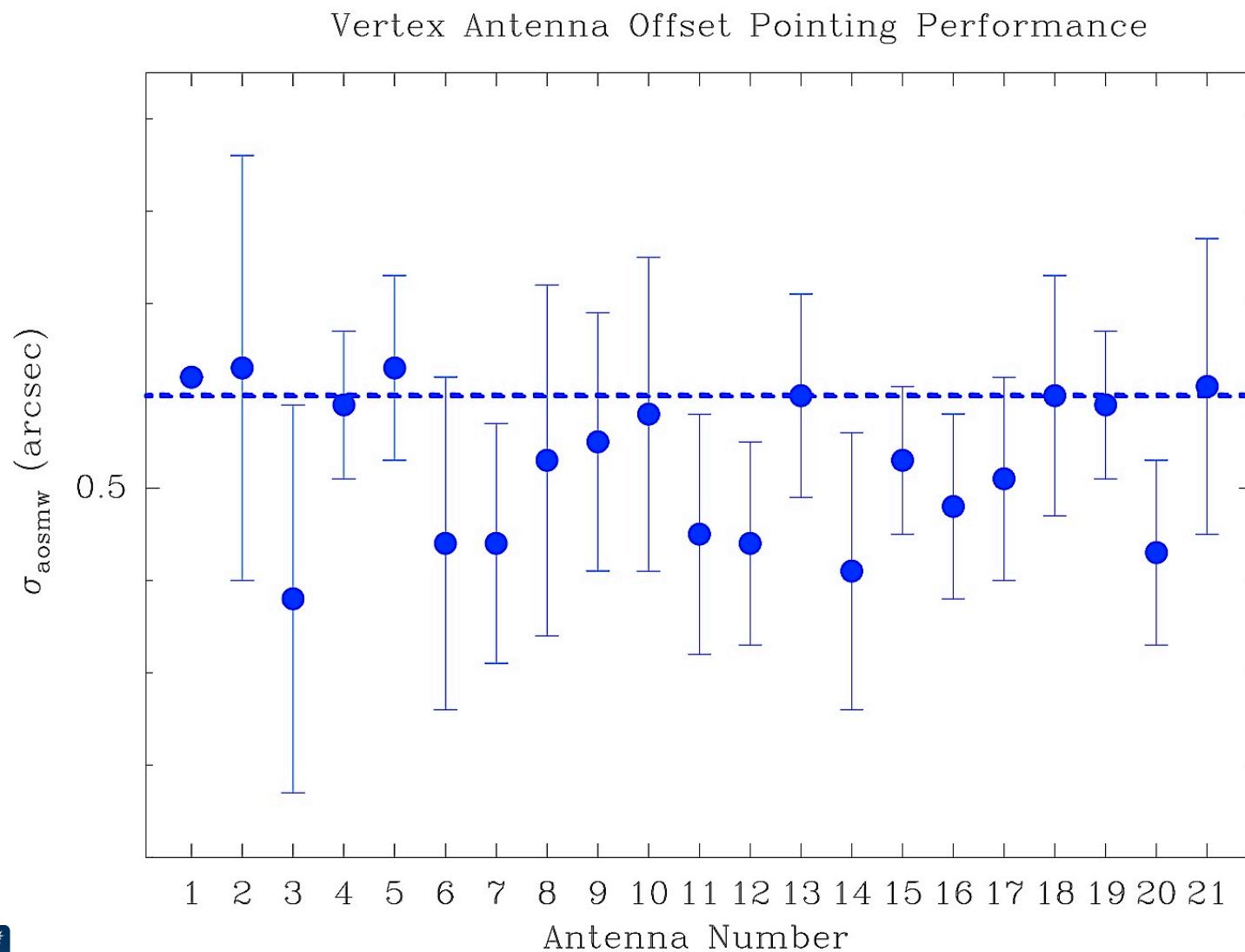


# Vertex Production Antenna Offset Pointing Performance

Antenna	Date <sup>a</sup>	N <sub>meas</sub>	T <sub>amb</sub> (C)	ΔT (C)	W <sub>s</sub> (m/s)	σ <sub>meas</sub> (arcsec)	σ <sub>aosmw</sub> (arcsec)
DV01	2009/02/02	40	+11.2 – +17.8	0.1 – 0.8	0.7 – 4.5	0.55±0.11	0.62±0.10
DV02	2009/04/15	29	+8.3 – +13.6	0.2 – 1.3	0.9 – 4.1	0.63±0.14	0.63±0.10
DV03	2009/10/05	36	+4.6 – +10.7	0.2 – 3.6	0.8 – 5.8	0.66±0.17	0.38±0.24
DV04	2010/02/03	30	+13.1 – +17.6	0.2 – 1.3	0.8 – 5.9	0.49±0.13	0.59±0.19
DV05	2009/10/26	30	+7.1 – +15.1	0.2 – 1.4	0.8 – 5.4	0.73±0.18	0.63±0.12
DV06	2010/06/10	32	+7.0 – +14.3	0.2 – 1.0	1.6 – 9.2	0.51±0.13	0.44±0.27
DV07	2010/04/16	26	+11.7 – +17.0	0.0 – 1.7	1.9 – 6.9	0.51±0.13	0.44±0.27
DV08	2010/07/03	26	+8.1 – +15.5	0.2 – 1.6	0.6 – 6.9	0.64±0.22	0.53±0.23
DV09	2010/08/02	27	+4.4 – +11.9	0.2 – 2.2	1.3 – 5.8	0.57±0.16	0.55±0.24
DV10	2010/10/13	25	+4.9 – +14.5	0.0 – 1.0	1.7 – 4.7	0.67±0.16	0.58±0.28
DV11	2010/11/29	28	+9.8 – +16.1	0.2 – 1.3	1.4 – 5.9	0.58±0.11	0.45±0.25
DV12	2011/01/30	25	+12.7 – +17.0	0.2 – 0.9	1.4 – 5.6	0.52±0.11	0.44±0.28
DV13	2011/03/23	32	+10.0 – +16.2	0.1 – 1.7	1.7 – 5.2	0.59±0.16	0.60±0.18
DV14	2011/05/31	54	+8.6 – +16.5	0.1 – 1.5	1.6 – 7.1	0.56±0.17	0.41±0.27
DV15	2011/07/29	45	+5.4 – +14.7	0.2 – 1.6	1.5 – 8.6	0.73±0.25	0.53±0.33
DV16	2011/08/27	40	+5.4 – 11.4	0.2 – 1.0	1.2 – 5.8	0.60±0.13	0.48±0.28
DV17	2011/10/10	30	+9.2 – +17.7	0.1 – 2.5	0.9 – 6.3	0.54±0.13	0.51±0.23
DV18	2011/11/28	27	+9.2 – +15.5	0.1 – 1.1	1.7 – 5.8	0.61±0.15	0.60±0.21
DV19	2012/01/14	29	+9.9 – +17.0	0.1 – 0.8	0.8 – 4.2	0.53±0.15	0.59±0.17
DV20	2012/04/07	32	+12.0 – +17.4	0.1 – 0.9	2.0 – 6.4	0.50±0.14	0.43±0.23
DV21	2012/05/04	29	+5.8 – +14.9	0.2 – 1.0	1.9 – 5.1	0.66±0.16	0.61±0.23

<sup>a</sup> Date of final report

# Vertex Production Antenna Offset Pointing Performance



# Things We Missed

- A “ghost deformation” due to “pinching” of the feet of the antennas
  - ✓ Interesting though negligible contribution to the all-sky pointing performance discovered while characterizing antenna DV06
- Azimuth encoder hysteresis
  - ✓ Encoder shaft alignment issue diagnosed and solved by antenna DV03
- Tiltmeter measurement dependence on local temperature
  - ✓ Flaw in metrology system design which only exposed itself during observations at the AOS (much wider range in ambient temperature)
  - ✓ Ethanol electrolyte freezing at low temperatures
  - ✓ Replacing methanol-based tiltmeters tested and found to work at low temperature



# Pointing Performance Summary

- Based on a limited set of positioning characterization measurements conducted at the OSF, all Vertex antennas meet the all-sky and offset pointing performance specifications
- Antenna pointing is a process which develops over the lifetime of an instrument



# Fast Switching

## Techniques and Systems Used

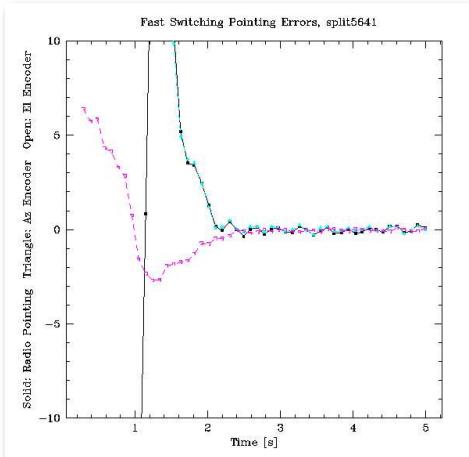
- Radiometer
- Optical Pointing Telescope
- Accelerometer System



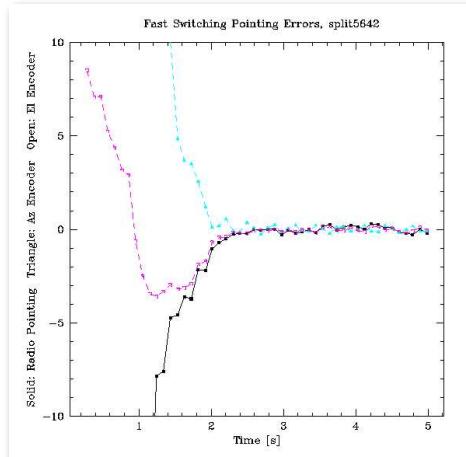
# Radiometric Fast Switching Performance

## Prototype Antennas

AZ

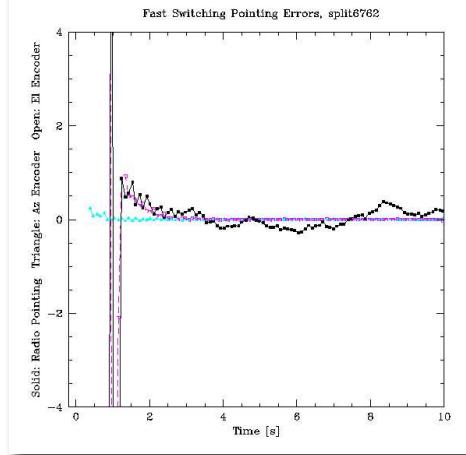
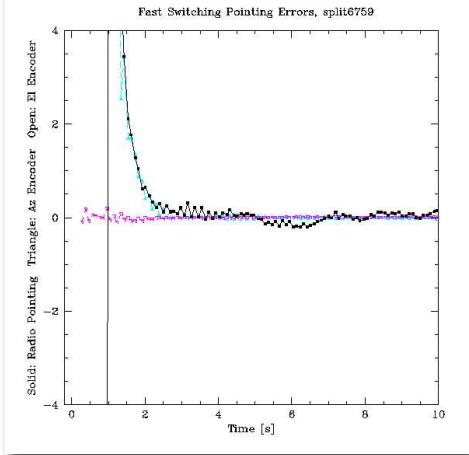


EL



VertexRSI

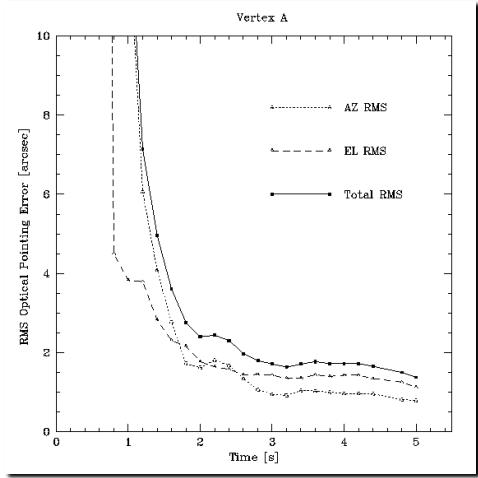
AEC



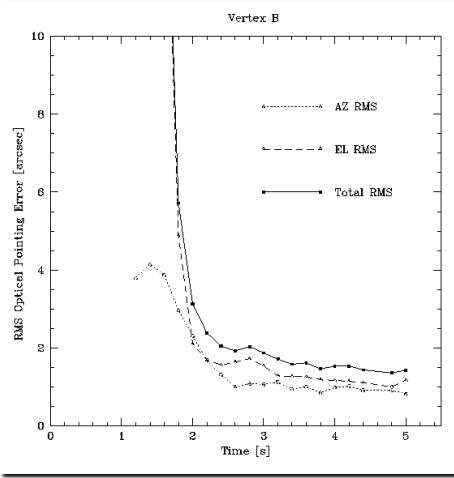
# Optical Fast Switching Performance

## Prototype Antennas

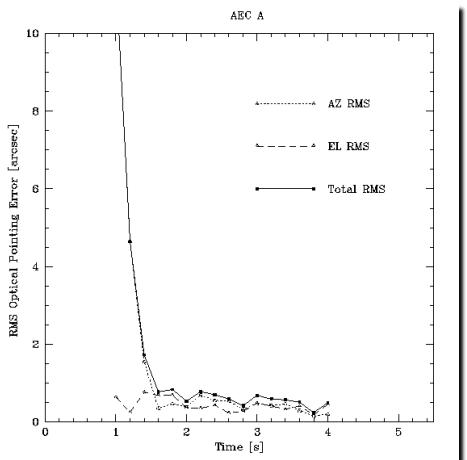
$$\theta_{\text{star}} = 0.4 \text{ deg}$$



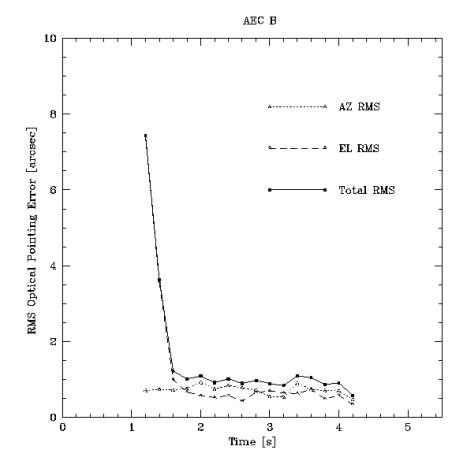
$$\theta_{\text{star}} = 1.8 \text{ deg}$$



VertexRSI



AEC



# Fast Switching Performance Summary

Prototype Antennas

<b><i>System/Technique</i></b>	<b><i>AEC</i></b>	<b><i>VertexRSI</i></b>
Radiometry	1.4-1.8 sec	1.5-1.8 sec
Optical	1.4-1.6 sec	1.5-2.0 sec
Accelerometer	1.5* sec	1.5-1.8 sec

\* Ignoring antenna drive shutdown due to apex oscillation



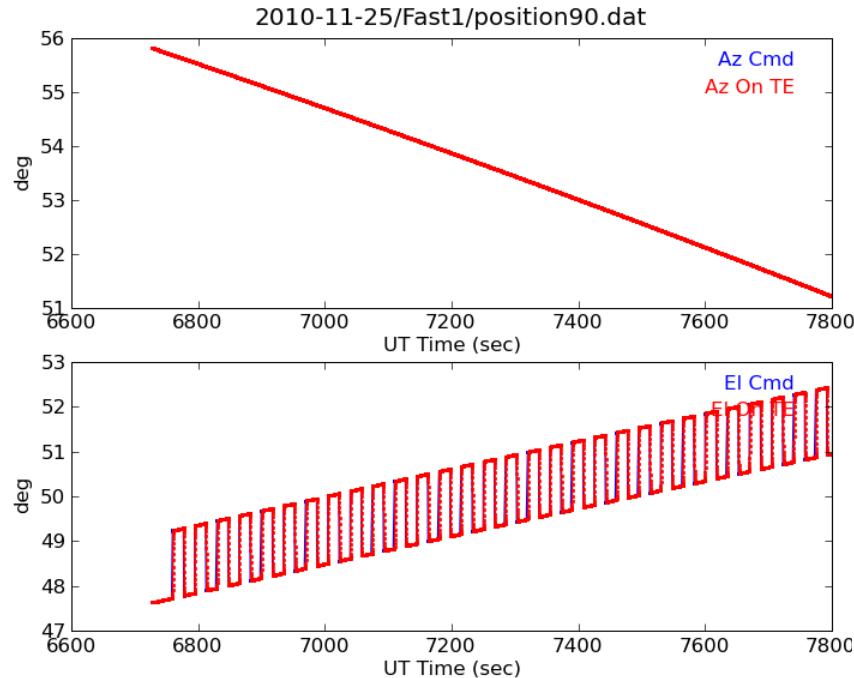
# Fast Switching Measurement and Analysis

## Production Antennas

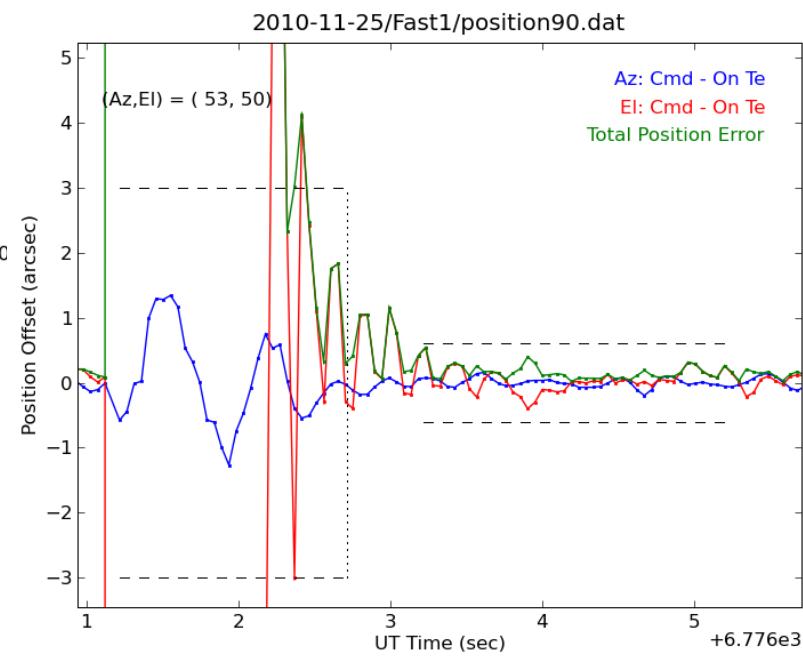
- Fast switch between two positions separated by 1.5 degrees
- Integrate at each position for  $\approx$  15 seconds
- Measure at (A,E) positions defined as follows:
  - A = 0-360 deg in steps of 60 deg
  - E = 30-60 deg in steps of 15 deg
  - PA = 270 (pure A switch), 0 (pure E switch), and 45 or 135 (mixed A/E switch) deg for each (A,E) matrix position
  - 3240 individual FS measurements spanning 54 FS runs, each with 60 FS measurements
- Fast switch for  $\approx$  10 minutes per (A,E) and position angle



# Fast Switching Measurements

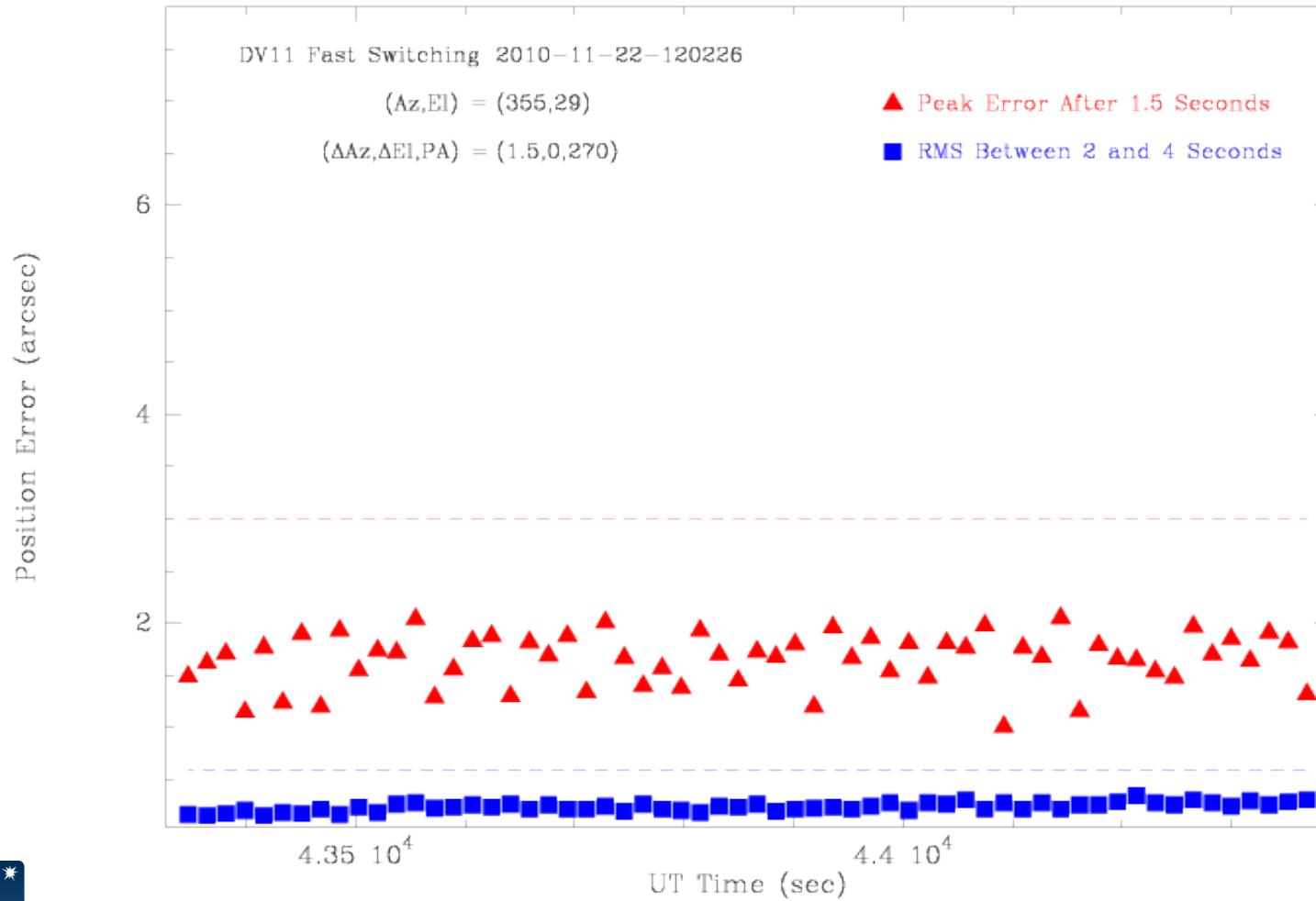


Production Antennas



# Example FS Measurement Run

## Production Antennas



# Fast Switching Performance

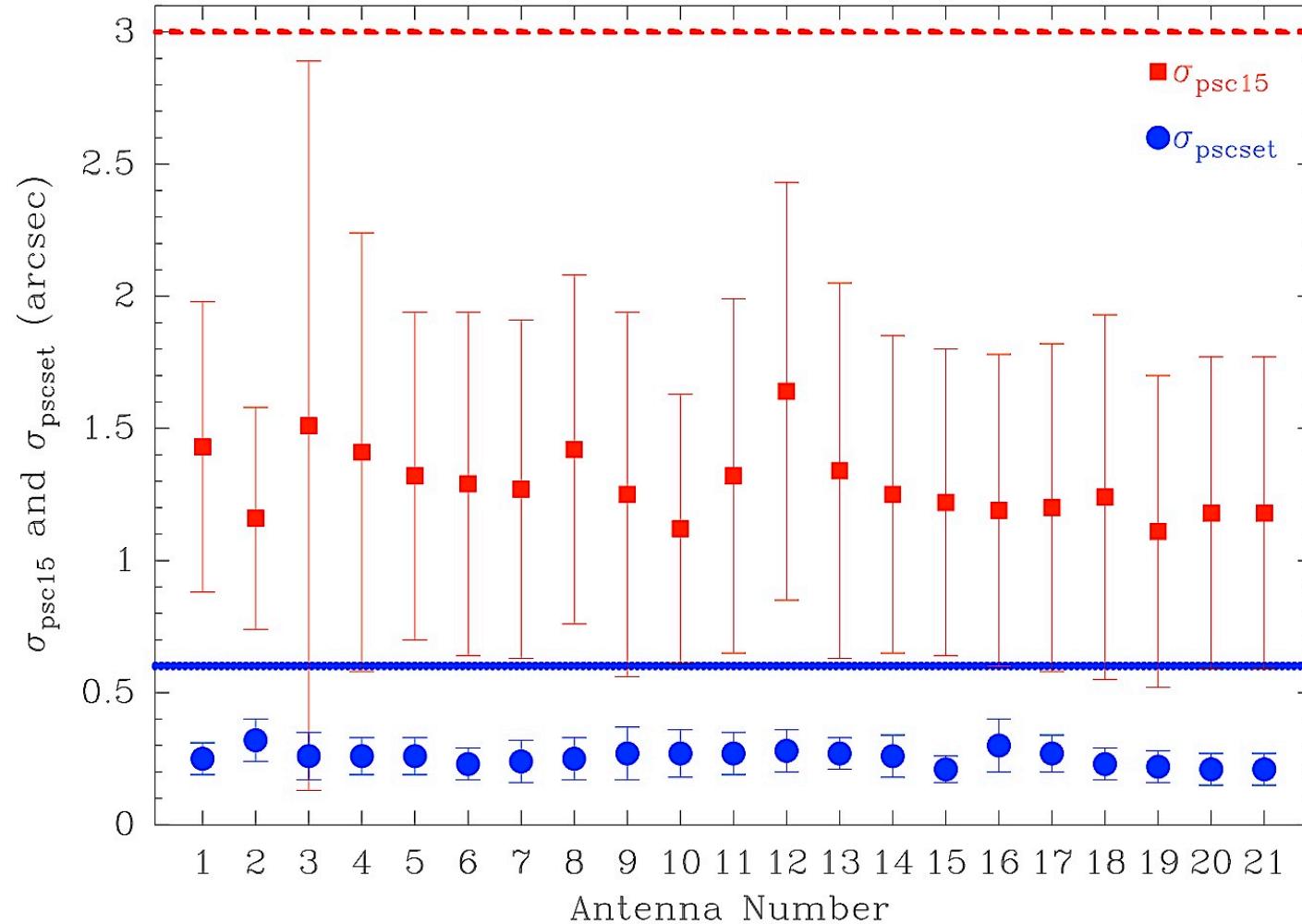
Antenna	Date <sup>a</sup>	N <sub>meas</sub>	T <sub>amb</sub> (C)	ΔT (C)	W <sub>s</sub> (m/s)	σ <sub>psc15</sub> (arcsec)	σ <sub>pscset</sub> (arcsec)
DV01	2009/02/10	660	+13.6 – +17.0	0.2 – 1.5	1.0 – 3.4	1.43±0.55	0.25±0.06
DV02	2009/04/28	1320	+8.5 – +12.9	0.5 – 1.8	0.8 – 3.2	1.16±0.42	0.32±0.08
DV03	2009/09/10	5150	+3.4 – +22.1	0.1 – 2.3	0.6 – 5.6	1.51±1.38	0.26±0.09
DV04	2010/02/01	3420	+12.5 – +23.7	0.1 – 1.7	0.5 – 9.2	1.41±0.83	0.26±0.07
DV05	2009/11/03	4620	+6.8 – +21.1	0.1 – 2.0	1.1 – 7.6	1.32±0.62	0.26±0.07
DV06	2010/05/09	3240	+7.2 – +17.6	0.3 – 2.5	0.4 – 4.7	1.29±0.65	0.23±0.06
DV07	2010/04/22	3240	+9.6 – +18.4	0.1 – 2.0	1.2 – 7.0	1.27±0.64	0.24±0.08
DV08	2010/05/23	3240	+6.4 – +17.9	0.4 – 2.0	0.2 – 6.7	1.42±0.66	0.25±0.08
DV09	2010/08/02	3240	+5.8 – +14.3	0.2 – 2.1	0.6 – 4.2	1.25±0.69	0.27±0.10
DV10	2010/10/11	3240	+5.7 – +18.2	0.0 – 1.9	1.3 – 5.3	1.12±0.51	0.27±0.09
DV11	2010/11/29	3240	+9.4 – +21.3	0.2 – 1.9	1.7 – 6.0	1.32±0.67	0.27±0.08
DV12	2011/01/22	3240	+13.0 – +20.5	0.1 – 2.0	0.7 – 4.3	1.64±0.79	0.28±0.08
DV13	2011/03/16	3240	+12.1 – +21.5	0.2 – 3.2	1.1 – 6.2	1.34±0.71	0.27±0.06
DV14	2011/05/30	3780	+9.6 – +18.0	0.1 – 1.4	0.6 – 4.5	1.25 0.60	0.26 0.08
DV15	2011/07/27	3420	+6.1 – +11.7	0.1 – 2.5	1.3 – 8.2	1.22 0.58	0.21 0.05
DV16	2011/08/26	3420	+7.0 – +16.3	0.1 – 1.8	1.1 – 5.9	1.19 0.59	0.30 0.10
DV17	2011/10/10	3240	+14.6 – +22.0	0.6 – 1.7	0.6 – 5.0	1.20 0.62	0.27 0.07
DV18	2011/11/27	3240	+10.7 – +20.3	0.1 – 2.4	0.5 – 7.5	1.24 0.69	0.23 0.06
DV19	2011/12/22	3240	+11.9 – +23.6	0.6 – 2.3	1.3 – 6.0	1.11 0.59	0.22 0.06
DV20	2012/03/28	3240	+12.4 – +21.9	0.3 – 1.6	0.9 – 3.7	1.18 0.59	0.21 0.06
DV21	2012/05/10	3240	+14.1 – +19.8	0.1 – 1.6	1.3 – 6.9	1.14 0.52	0.23 0.06

<sup>a</sup> Date of final report

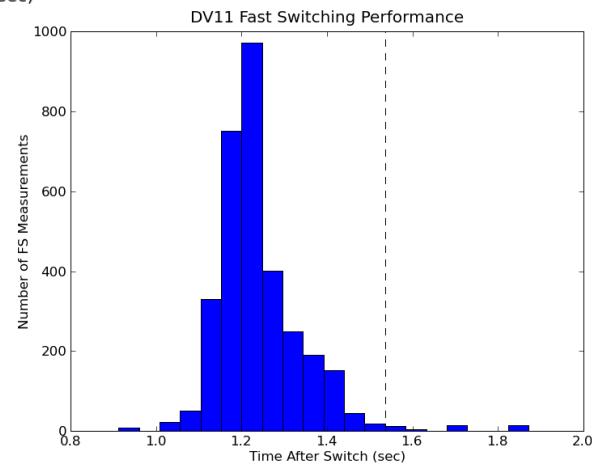
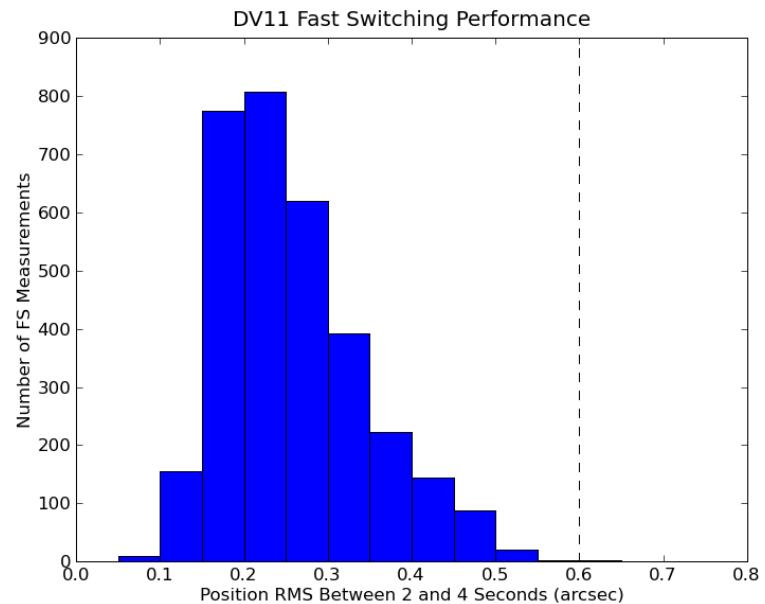
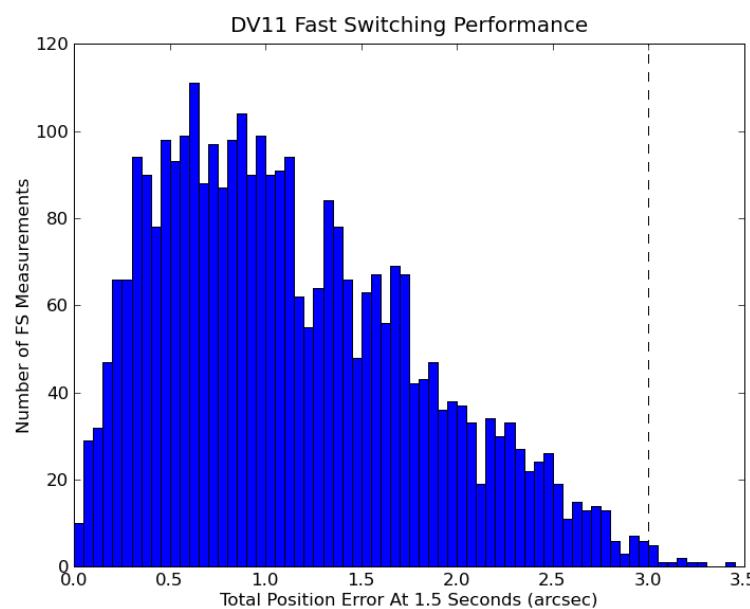
# Fast Switching Performance

## Production Antennas

Vertex Antenna Fast Switching Performance

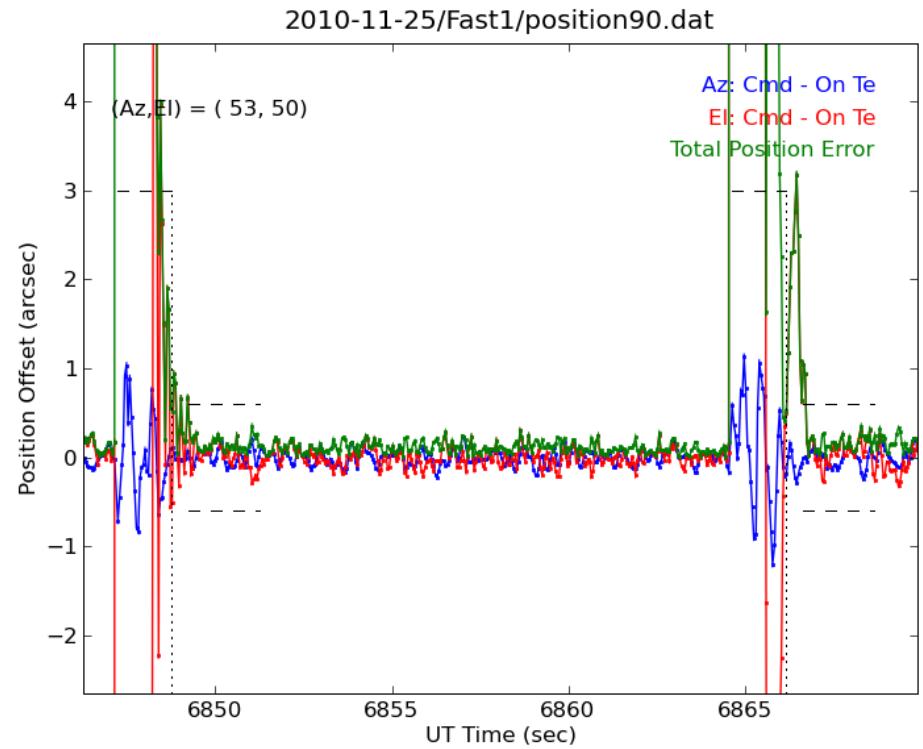
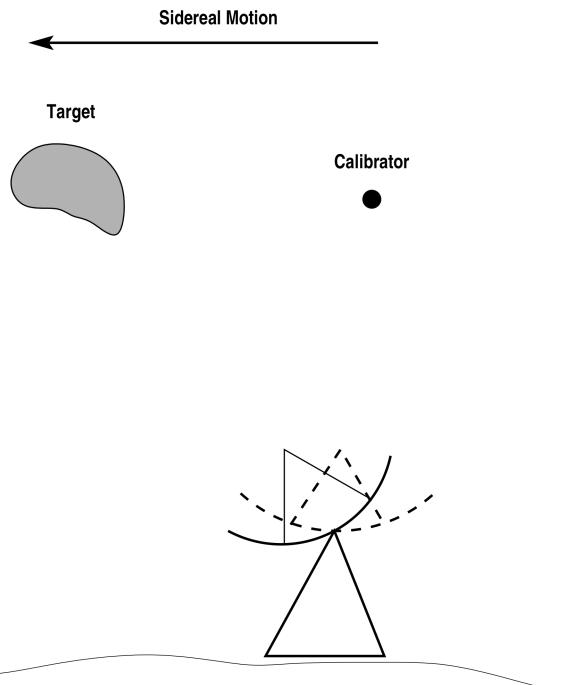


# Sample DV Fast Switching Performance



# Differential Sidereal Motion and FS

- During the time that it takes to move from one position to another 1.5 degrees away the Earth has rotated a bit.
- In certain regions of the sky this differential sidereal motion can add a bit of distance to a nominal 1.5 degree fast switch.

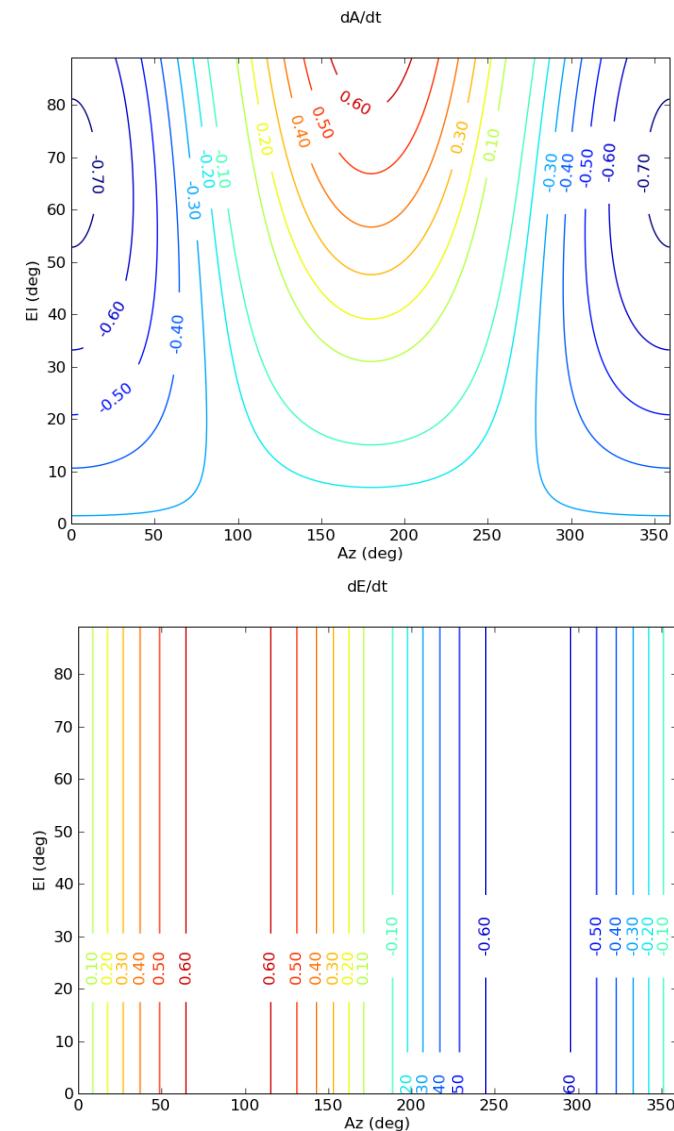
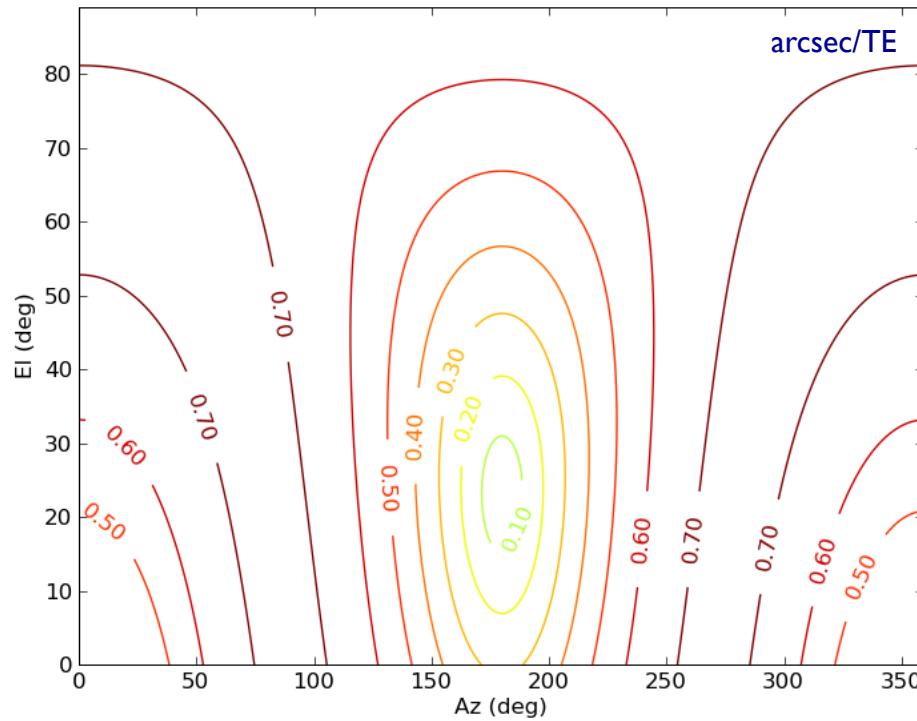


# Differential Sidereal Calculation

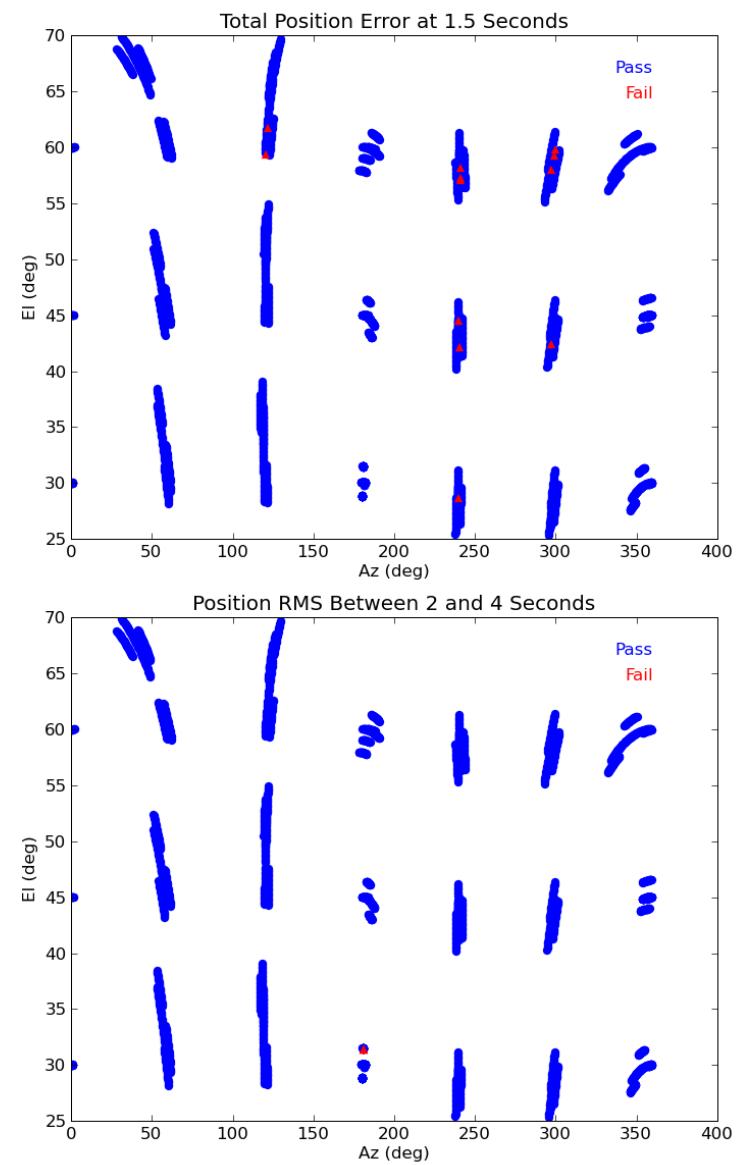
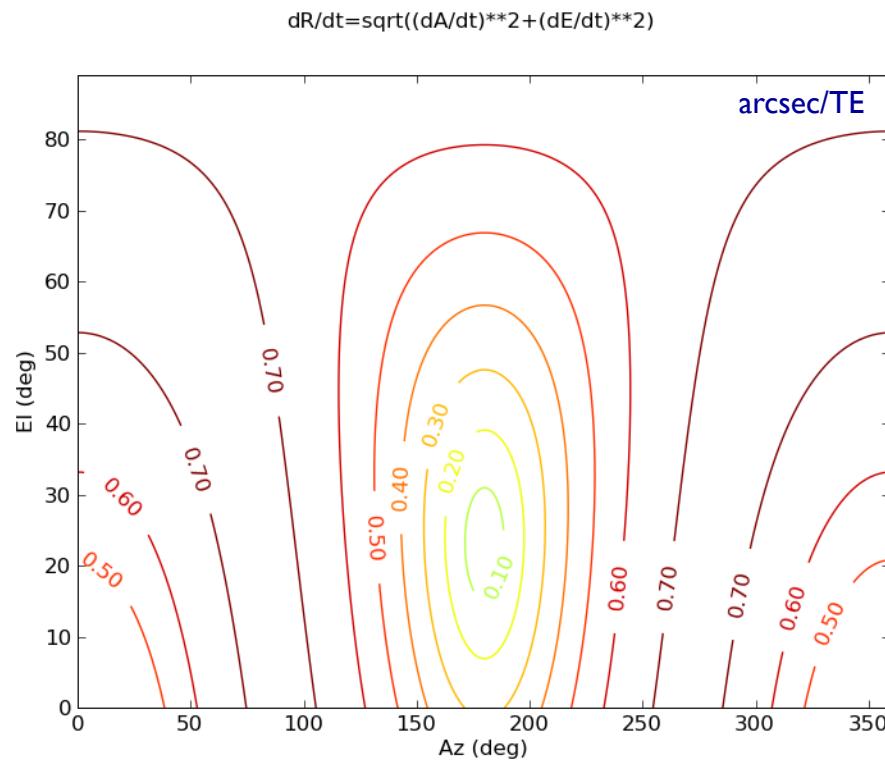
$$\frac{dA}{dt} = \frac{d\tau}{dt} \left[ \frac{\sin \phi \cos E - \cos \phi \sin E \cos A}{\cos E} \right]$$

$$\frac{dE}{dt} = \frac{d\tau}{dt} \cos \phi \sin A$$

`dR/dt=sqrt((dA/dt)**2+(dE/dt)**2)`

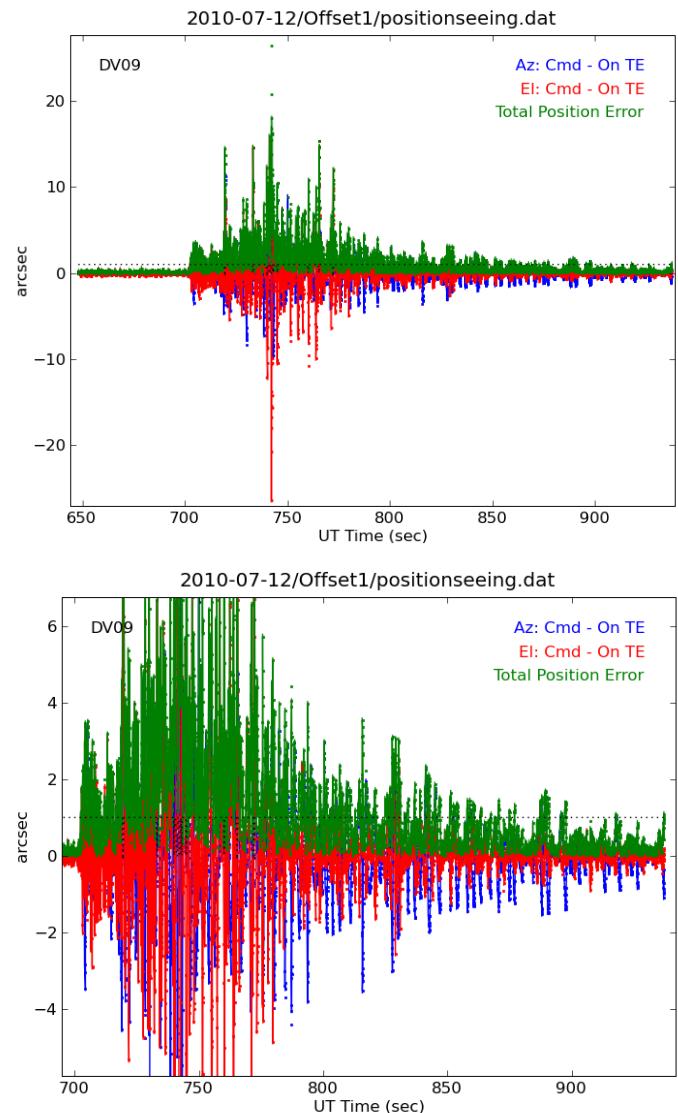


# Differential Sidereal Affect on FS Measurements



# A Multi-Million Dollar Earthquake Detector

- On the evening of July 12 2010 a magnitude 6.2 earthquake jolted a remote area approximately 80 km ENE of Calama (latitude =  $-22.26$ , longitude =  $-68.20$ ; 0.86 degrees of distance on the earth's surface).
- When the earthquake started DV09 was making a PSC measurement as part of a regular seeing observation.
- Operator noted: "The star did figure-eights on the screen for about 3 minutes".
- Using information from the USGS listing for this earthquake at <http://earthquake.usgs.gov/earthquakes/recenteqsww/Quakes/us2010yqad.php>:
  - ✓ The PSC diagram shown indicates that the quake began at the ALMA site at UT 701 seconds, which is 00:11:41 UT.
  - ✓ The USGS lists the start time of the quake as 20:11:18 local time, which is 00:11:18 UT.
  - ✓ The USGS web site has an online propagation time calculator which allows one to calculate the propagation time from the epicenter to the ALMA site.
  - ✓ The estimated propagation time is about 18 seconds, which is tantalizingly close to the 23 second difference between the USGS quake time of 00:11:18 UT and the measured start time of 00:11:41.
  - ✓ Propagation time from the epicenter to the ALMA site is the explanation for the difference in measured start times.
- The quake shook the antenna pretty hard for about 100 seconds, then slowly damped until well after the end of the PSC measurements.
- The PSC measurements ended around UT 938 seconds, or 00:15:38 UT, while the antenna was still being shaken (see lower panel).
- The antenna reacted to the earthquake for more than 4 minutes.
- We note this event due to the fact that two all-sky pointing runs were made just before the earthquake, while offset pointing measurements 001751 and 004516 were made immediately after this event.
- None of these measurements show any ill effects due to the earthquake.



# Fast Switching Performance Summary

- All Vertex antennas meet both parts of the fast switching specification



# Reflector Surface Accuracy

Techniques and Systems Used:

- Near-Field Radiometric Holography
- Accelerometer System
- Radiometric Efficiency Measurements
- Temperature Sensor System



# Near-Field Radiometric Holography System



- Transmitter:
  - Photonically-generated signal
  - 50m high tower located 310m from antenna
- Receiver:
  - 79/104 GHz dual-receiver system
  - 10  $\mu\text{m}$  spec; 5  $\mu\text{m}$  goal
  - Gildas data analysis (Lucas)

# Surface Setting

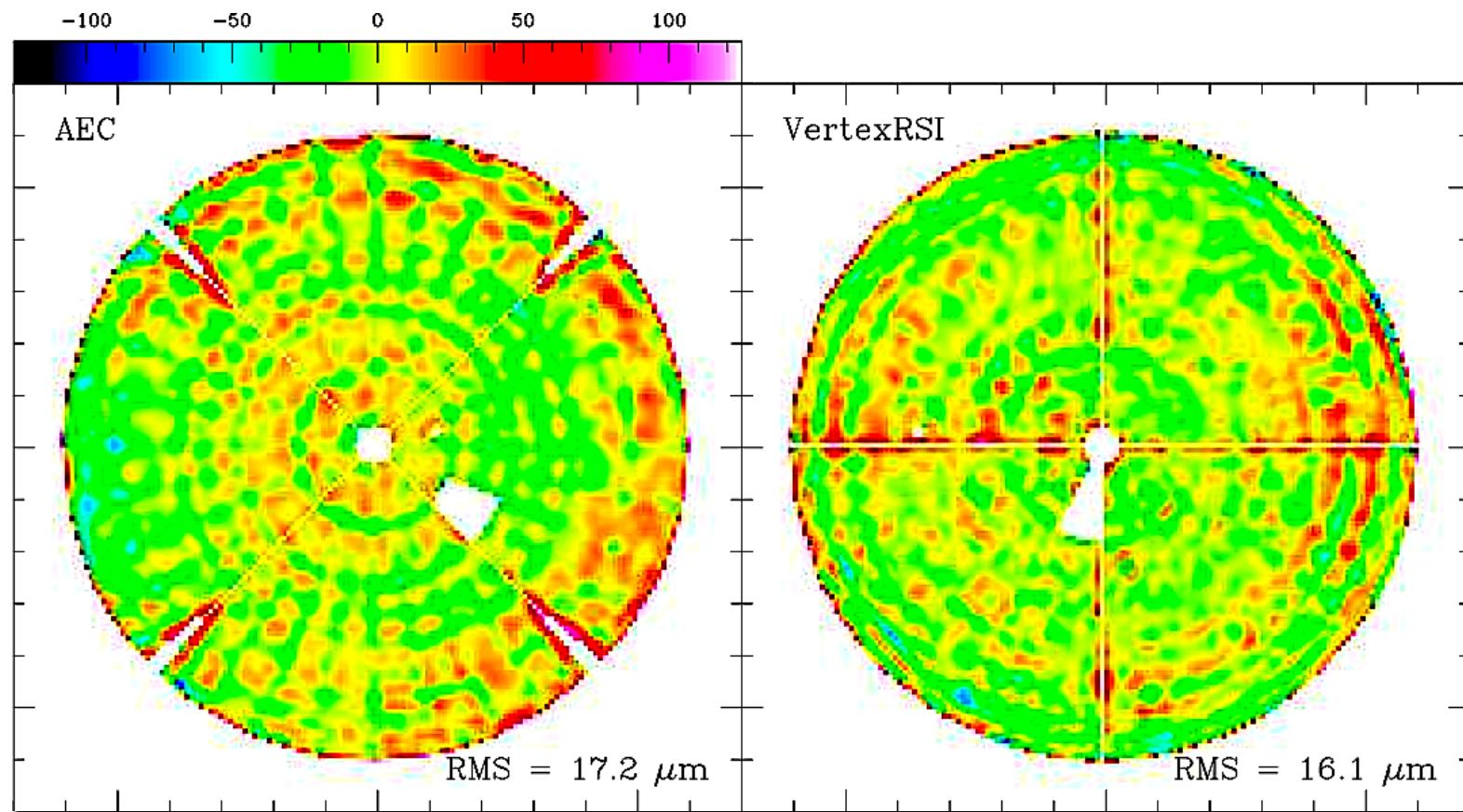


Prototype Antennas



# Near-Field Radiometric Holography Results

Prototype Antennas



# Reflector Surface Accuracy Performance Summary

Prototype Antennas

<b><i>Technique/System</i></b>	<b><i>AEC</i></b>	<b><i>VertexRSI</i></b>
Holography	$17 \pm 5 \mu\text{m}$	$16 \pm 5 \mu\text{m}$
Accelerometers	$\Delta\text{RMS} < 8 \mu\text{m}$	$\Delta\text{RMS} < 7 \mu\text{m}$
Radiometric	$< 40 \mu\text{m}$	$< 40 \mu\text{m}$
Temperature Sensors	...	$\Delta\text{RMS} \leq 3 \mu\text{m}$



# Near-Field Radiometric Holography Results

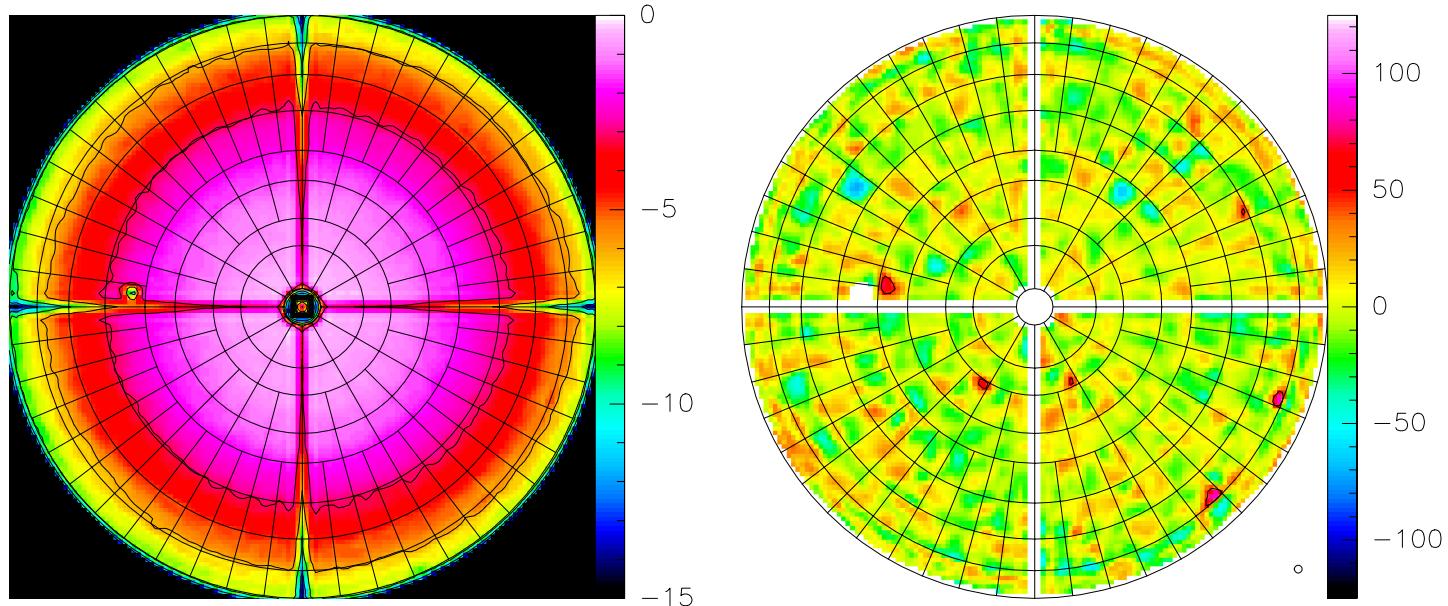
Production Antennas

```

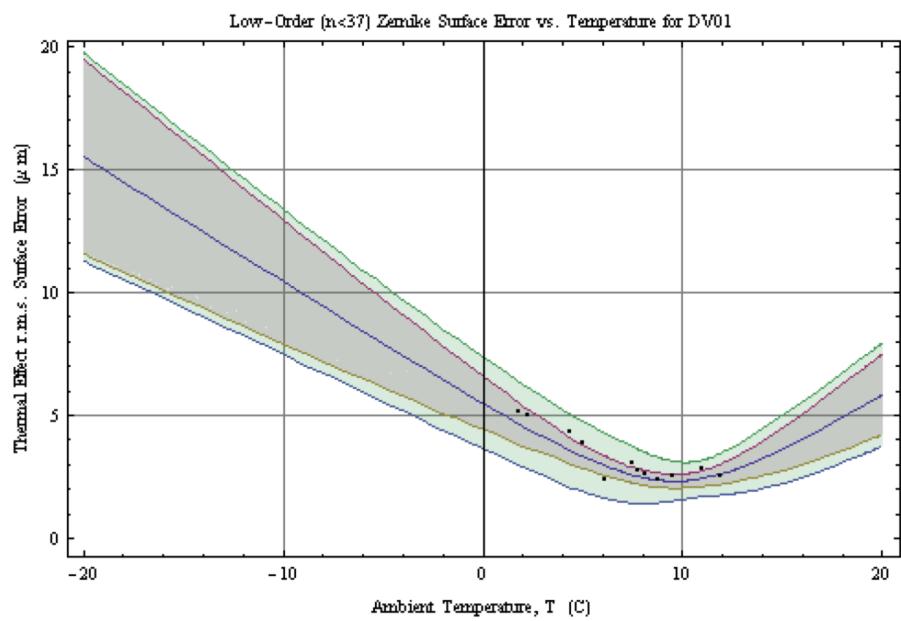
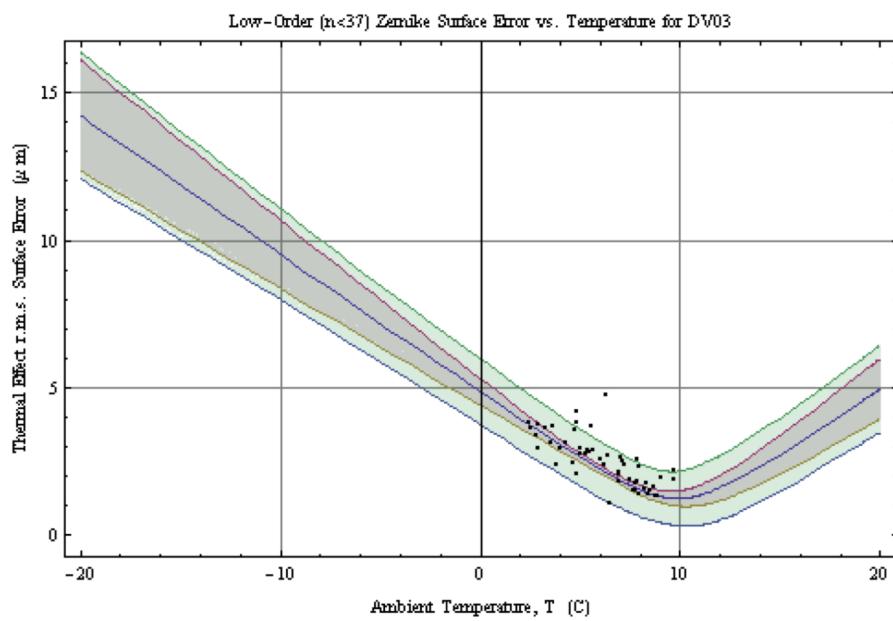
RF: Umaplc - 14-JUL-2009 23:22:30 - almaproc@oper03 - ALMA03 - ALMA/Vertex 12-m Pro
Am: Rel.(B)   ATFTower    test      scans  2 to 216 14-JUL-2009 11:31UT El: 9.75
Ph: Rel.(B)

rms Pha.          Edge taper = 15.61x 15.46 dB - offset X= -0.15 Y= -0.02 m
12  0.110        Focus offsets (X,Y,Z) = 4.65 -2.49 7.56 mm; Astigmatism = 0.00 mm
                           Phase rms (unweighted)= 0.060 (weighted)= 0.059 radians
                           Surface rms (unweighted)= 13.70 - (weighted)= 13.51 μm
 $\eta_A(104.020 \text{ GHz}) = 0.815; \eta_A(230.0 \text{ GHz}) = 0.804; \eta_A(345.0 \text{ GHz}) = 0.787$ 
S/T(104.020 GHz)= 29.939 Jy/K; S/T(230GHz)= 30.345 Jy/K; S/T(345 GHz)= 30.990 Jy/K
 $\eta_I = 0.818 -\eta_S = 0.831 -\eta_P(104.020 \text{ GHz}) = 0.997 -\eta_P(230 \text{ GHz}) = 0.983 -\eta_P(345 \text{ GHz}) = 0.963$ 
Rms/ring: 13.0 12.9 12.9 14.3 12.1 13.5 14.5 14.2
Amplitude (front view)           Normal errors (front view)
-15.000 to 0.000 by 3.000      -125.000 to 125.000 by 50.000

```



# ALMA Vertex Production Antenna Measured Temperature Dependence



# Gravitational and Thermal Deformation

rlucas@gns 09-JUN-2010 20:25:50

ALMA

Result file: Diff-0-90.map

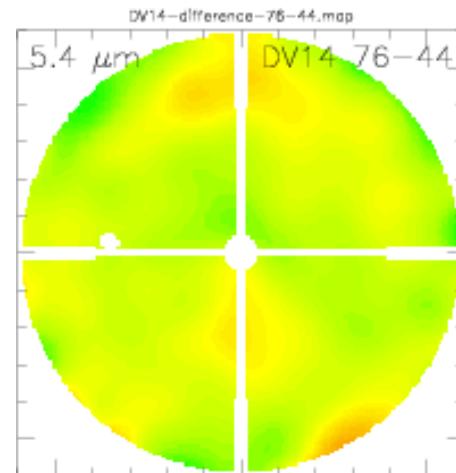
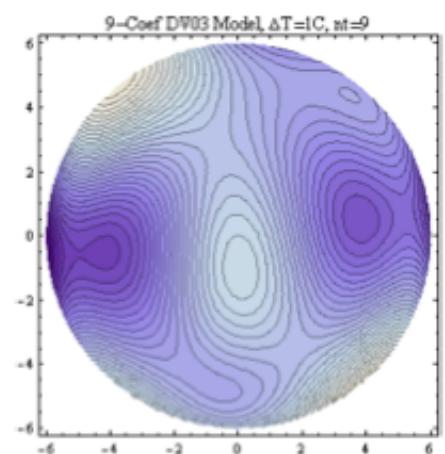
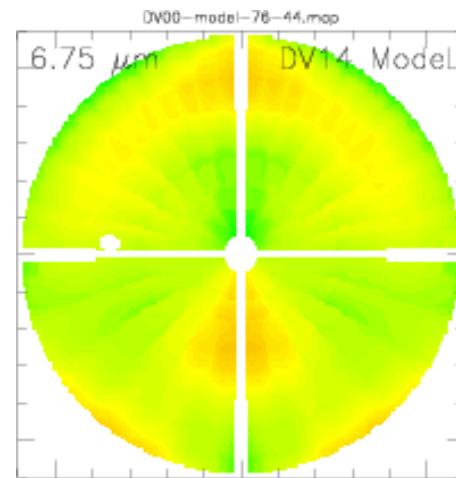
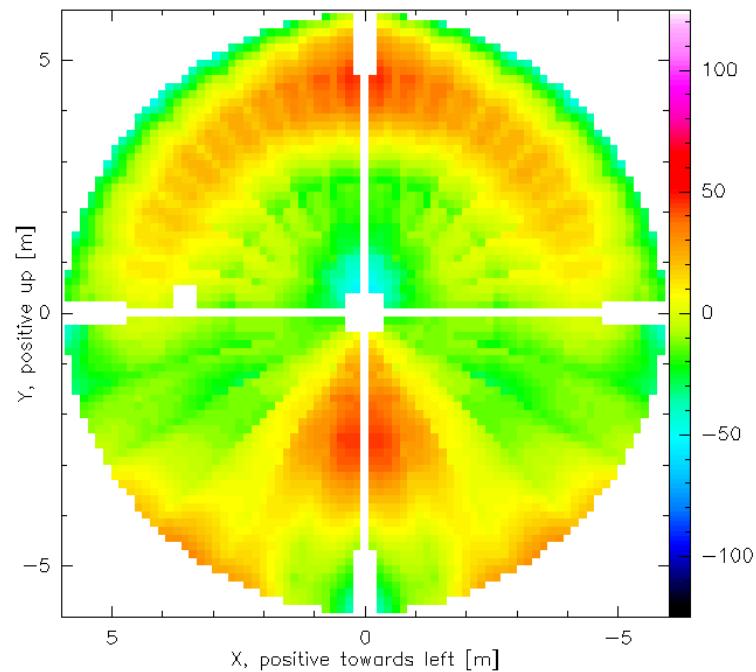
- Required panel motion towards focus ( $\mu\text{m}$ )  
(positive number means a hole)
- difference map as seen from focus.

/users/rlucas/ASTROHOLO/DV01 DV02 DV03 DV05 PM03/20100604/

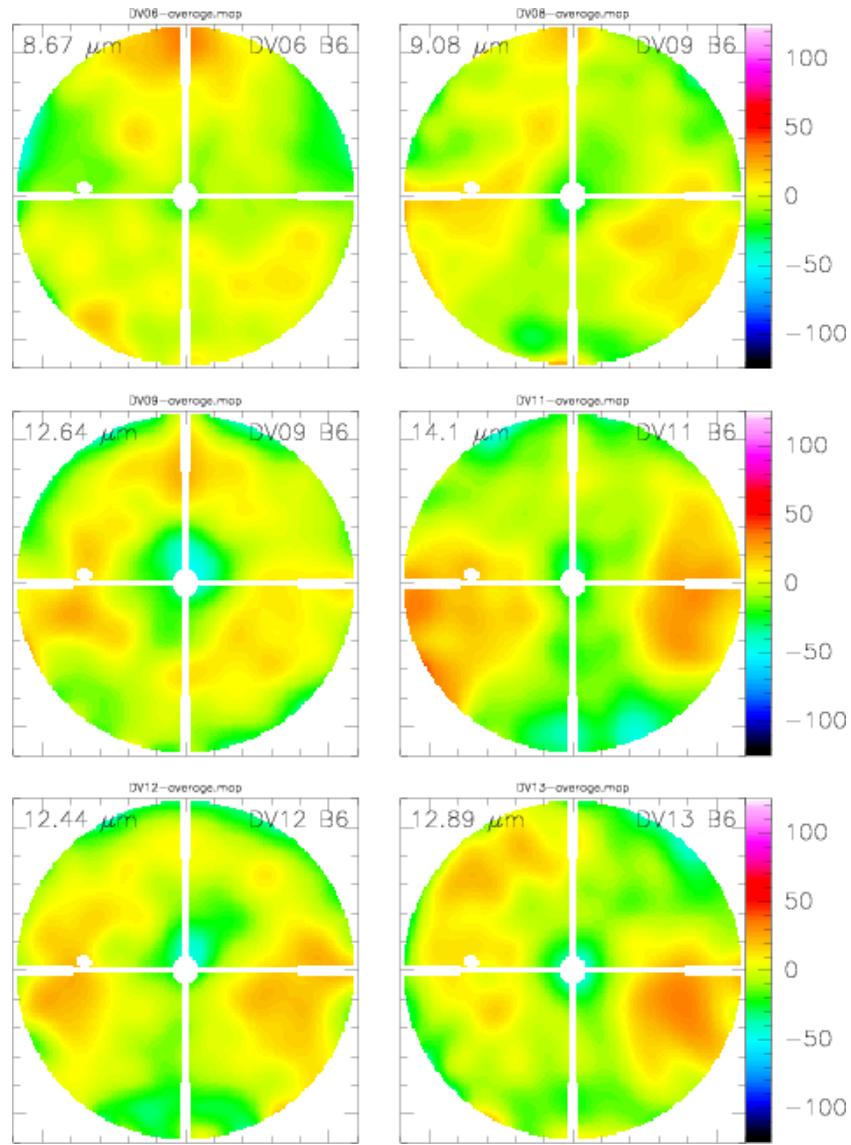
/users/rlucas/ASTROHOLO/DV01 DV02 DV03 DV05 PM03/20100604/  
rms (unweighted) 17.39  $\mu\text{m}$

rms (amp. weighted) 17.24  $\mu\text{m}$

rms (12dB weighted,  $\cos \alpha$  included) 16.35  $\mu\text{m}$



# ALMA Vertex Production Antenna Astronomical Holography



# Publications

- Mangum, J. G., Baars, J.W. M., Greve, A., Lucas, R., Snel, R., Wallace, P.T., & Holdaway, M. 2006, PASP, 118, 1257: ***Evaluation of the ALMA Prototype Antennas.***
- Baars, J.W. M., Lucas, R., Mangum, J. G., & Lopez-Perez, J. 2007, IEEE Antennas and Propagation Magazine, 49, 24: ***Near-Field Radio Holography of the Large Reflector Antennas.***
- Snel, R. C., Mangum, J. G., & Baars, J.W. M. 2007, IEEE Antennas and Propagation Magazine, 49, 84: ***Study of the Dynamics of Large Reflector Antennas with Accelerometers.***
- Greve, A. & Mangum, J. G. 2008, IEEE Antennas and Propagation Magazine, 50, 66: ***Mechanical Measurements of the ALMA Prototype Antennas: Path Lengths, Thermal Behaviour, and Azimuth Bearing.***



# Summary

- The ALMA prototype and production antennas meet or surpass all performance metrics:
  - Positioning (pointing, fast switching, OTF, etc.)
  - Surface Accuracy
  - Path Length Stability
- Large (12m) efficient antennas can be built allowing precision radiometry up to 1 THz
- Many aspects of the ALMA antenna performance (i.e. stability with time, transport, environment, etc.) still need to be studied
- The effort expended in developing high-performance antennas always results in observatory efficiency and longevity

